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<u>Item</u>	Page	
Summary	1 1/	A7
Introduction	2 1/	A8
Experimental Apparatus	4	1/A1
Initial Experiments Using Piezoelectric Transducers	4 1/	A10
Final Experimental Arrangement	4 1/	A10
Circular Cylinder Models		A11
Instrumentation	8	A14
Microphone Calibration	8 1/	A14
Discussion of Results	12 1/	В6
Surface Pressure Augmentation	12 1/	В6
Phase Angle Variations	18 1/	B12
Scattering Parameter	19 1/	B13
Conclusions	31 1/	C11
References	32 1/	C12
Appendices	33 1/	C13

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Acoustic Scattering by Circular Cylinders of Various Aspect Ratios

Algirdas Maciulaitis

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Acoustic Scattering by Circular Cylinders of Various Aspect Ratios

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Prepared for Langley Research Center under Contract NAS1-14766



Scientific and Technical Information Office

1979

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LIST OF ILLUSTRATIONS

Figure		Page
1	Cylindrical Model and Loudspeaker inside the Anechoic Room	6
2	Scattering Model	6
3	Interior of the Cylindrical Model	7
4	Loudspeaker Adaptor Bracket for Alignment with the Laser Gun	7
5	Microphone and Preamplifier Installation Inside the Cylindrical Model	9
6	Data Acquisition Circuit Schematic	10
7	Computer and Some of the Instrumentation Used in the Scattering Experiments	11
8	Schematic Plan View of the Scattering Experiment Geometry	14
9	Surface Pressure Augmentation on an L/D = 0.5 Cylinder Body at ka = 0.407	15
10	Surface Pressure Augmentation on an L/D = 0.5 Cylinder Body at ka = 2.288	16
11	Surface Pressure Augmentation on an L/D = 0.5 Cylinder Body at ka = 4.232	17
12	Surface Pressure Augmentation on an L/D = 0.25 Cylinder Body at ka = 0.407	21
13	Surface Pressure Augmentation on an L/D = 0.25 Cylinder Body at ka = 2.288	22
14	Surface Pressure Augmentation on an L/D = 0.25 Cylinder Body at ka = 4.232	23
15	Pressure Prase Variations on an L/D = 0.5 Cylinder Body at ka = 0.407	24
16	Pressure Phase Variations on an L/D = 0.5 Cylinder Body at ka = 2.288	25
17	Pressure Phase Variations on an $L/D = 0.5$ Cylinder Body	26

Figure			Page
18	Pressure Phase Variations on an $L/D = 0.25$ Cylinder at ka = 0.407		 27
19	Pressure Phase Variations on an $L/D = 0.25$ Cylinder at ka = 2.288		 28
20	Pressure Phase Variations on an L/D = 0.25 Cylinder at ka = 4.232	-	 29
21	Variation of the Scattering Parameter as a Function Circular Cylinders Having L/D of 0.5 and 0.25		30

SUMMARY

A frequently used configuration for pressure gradient (PG) microphones, designed to measure the spatial acoustic pressure variation (gradient) at a point in space, is a short circular cylinder having either two pressure transducers flush-mounted into the ends of the cylinder body, or a single differential transducer measuring pressure difference between cylinder ends.

Acoustic scattering on a microphone body can severely limit the useful frequency range of pressure gradient microphones. These scattering effects were investigated experimentally between ka values of 0.407 and 4.232 using two circular cylindrical models (L/D = 0.5 and 0.25) having a 25 cm outside diameter. Small condenser microphones, attached to preamplifiers by flexible connectors, were installed from inside the cylindrical bodies. A 38 cm diameter woofer in a large speaker enclosure was used as the sound source. The experiment was done in the new anechoic chamber at the NASA Langley Research Center. Sound waves were not assumed to be plane.

Surface pressure augmentation and phase differences were computed from measured data for various sound wave incidence angles. Results are graphically compared with theoretical predictions supplied by NASA for ka = 0.407, 2.288, and 4.232. All other results are tabulated in the appendices. With minor exceptions, the experimentally determined pressure augmentations agreed to within 0.75 dB with theoretical predictions. The agreement for relative phase angles was within 5 percent without any exceptions. This is excellent, and approaches the realistic repeatability limits in an acoustic experiment of the type reported here. The fact that such agreement was achieved means that the theoretical procedure is fully validated and can be used in investigating, with confidence, scattering about any axisymmetric shape. It also means that the experimental technique employed possesses the necessary precision to explore acoustic scattering situations where a theoretical analysis might not be feasible at the present time.

Scattering parameter variations with ka and L/D ratio, as computed from experimental data, are also presented. This type of data represents a useful tool in the design of pressure gradient microphones.

INTRODUCTION

As its name implies, the function of a pressure gradient (PG) microphone is to measure the slope in the spatial acoustic pressure variation. In practice, the slope is determined from simultaneous acoustic pressure measurements using either two back-to-back mounted pressure transducers, or a single transducer exposed to the ambient pressure on both sides. A common geometric configuration for a PG microphone is a short circular cylinder. PG microphones are used in conjunction with ordinary pressure (P) microphones to study the acoustic source details by taking measurements in the far field. They can also be used to measure the acoustic intensity vector. In this application, the particle velocity is obtained from the local pressure gradient by means of the momentum equation.

In designing pressure gradient microphones for maximum frequency response, it is mandatory that the effect of the presence of the microphone body on the measurables be fully taken into account. Only then is it possible to design a PG microphone that, depending on the operating frequencies, is either practically distortion free, or if not, can still be utilized successfully if proper corrections are made. Such corrections can be determined from preliminary investigations of the type described in this report. As far as it could be determined, the first systematic theoretical effort to optimize body shapes for reducing the effect of body scattering on pressure gradient microphone frequency response was reported in Ref. 1. The present study was aimed at an experimental verification of the findings of that work.

The main reason for the microphone disturbance of the acoustic pressure field existing in the absence of the microphone body is due to acoustic scattering by the body surface. Until recently, scattering fields could be computed only for simple geometric shapes whose surfaces constitute a coordinate surface in a coordinate system in which the wave equation is separable. The oblate spheroids constitute one such family of body shapes. This was the reason prompting the selection of that particular body shape as the scattering model in Ref. 2. The idea there was to confirm experimentally the scattering pressure distributions generated by a computer program. Having demonstrated very good agreement between experiment and

theory in that instance, it was decided to extend the proven experimental technique to bodies of revolution holding a greater practical interest from a point of view of application in PG microphones; namely circular cylinders of various length-to-diameter ratios. At the time the present work was initiated, no theoretical scattering solutions for this geometric shape were available. The situation has changed in the interim, and the present report contains numerous comparisons between experimental and analytical results.

Our approach in the experiments is to make the scattering models large (25 cm diameter) and hollow, which brings about several advantages. The main advantage of the size is that it affords a reasonable surface spatial resolution with use of conventional transducers that have a high pressure sensitivity. The additional advantage accrued consists in our ability to mount the transducers from inside the body and to run the electrical conduits through the model support pipe. This results in a very clean configuration, which in its "lollypop" shape closely resembles a PG microphone.

To maintain a uniform incident spherical sound field, tests were conducted inside an anechoic chamber. Although during the course of the contract work models of various L/D ratios were designed and fabricated, some having various types of edge roundness, lack of time and funds allowed testing only with two square-edged circular cylinder models having L/D of 0.5 and 0.25. It has been shown recently (in Ref. 1) that L/D = 0.5 is about the optimum aspect ratio for a PG microphone and that the effect of rounding the edges on the usable frequency range is minimal. In view of these findings, the two models tested do not constitute an overly restricted range of variables.

The data acquisition procedure, as finally adopted, was fully automated and was under computer control. This included changes in frequency, incidence angle, multiplexed measurements, and averaging of six different pressure amplitudes and five relative phase angles. John M. Seiner of the NASA Langley Research Center prepared the computer program that controlled the experimental data acquisition.

EXPERIMENTAL APPARATUS

INITIAL EXPERIMENTS USING PIEZOELECTRIC TRANSDUCERS

Our original plan was to conduct all scattering experiments in the anechoic chamber of the Pennsylvania State University. The chamber and the loudspeaker were tested using a standard condenser microphone mounted in a two-dimensional transversing mechanism. These tests helped to decide on proper locations for both the loudspeaker and the scattering models, and which frequencies would be particularly suited for scattering experiments.

(Within a desired frequency operating range these frequencies depend on the loudspeaker characteristics and on room acoustics.)

In the original configuration, scattering models were instrumented with piezoelectric type pressure transducers. Unfortunately, after extensive tests, it was realized that these transducers were unsuitable because, owing to the scattering model design and its support within the anechoic chamber, slight mechanical vibrations were induced in the scattering model when the loudspeaker output power was raised to a sufficiently high level to produce reasonable signal levels from the piezoelectric transducers. Under these conditions the transducers began acting as accelerometers and the acoustic signal was no longer recognizable. It was decided to switch to condenser microphones as pressure transducers. This meant added design and machining effort. Unavoidably, this change caused some delays and a certain amount of duplication of effort.

FINAL EXPERIMENTAL ARRANGEMENT

The experiment was ultimately conducted in the new anechoic chamber at the NASA Langley Research Center. The inside dimensions of the chamber are 3 by 4 by 2.5 m (2.5 being the height). Preliminary tests showed the chamber to be spechoic down to below 178 Hz, the lowest frequency used in the experiments.

The tests consisted of measuring the surface pressures and phase angles on two circular cylindrical bodies (L/D of 0.5 and 0.25) exposed to the

harmonic sound field emanating from a 38 cm diameter lead guitar speaker mounted in a 100 Hz enclosure. The distance between the loudspeaker and the center of the cylinder model was 1.35 m. A photograph of the loudspeaker and a cylindrical model mounted inside the anechoic chamber is shown in Fig. 1.

Circular Cylinder Models

A simplified drawing of the scattering model is shown in Fig. 2. Two circular cylinder models were used in the tests. They both had an outside diameter of 25 cm, but different L/D ratios: 0.5 and 0.25. These particular L/D ratios were selected because of the strong indication in Ref. 1 that 0.5 is about the optimum L/D ratio for a PG microphone. The models were machined from aluminum and consisted of six major parts each. Two transducer (microphone) holders were each mounted in an end plate that could be rotated around the cylinder axis. The cylinder body itself was made of two parts to provide access to the interior. One of the microphone holders had provisions to hold five 0.635 cm diameter microphones arranged on a radial line at 2.46 cm centers, starting with one microphone on the cylinder axis. The other microphone holder, which was installed in the end plate at the opposite end of the model, contained only a single microphone on the cylinder axis. Figure 3 shows a photographic view of the inside of the cylindrical model, which also shows the special adaptors for flush mounting the microphone cartridges.

The model was supported in the anechoic room on a model holder machined from a 3.175 cm diameter pipe. Microphone cables were run through the inside of this holder. The holder itself was attached to an in-line small electric stepping motor to provide accurate rotational positioning of the model about a vertical axis.

To ensure proper alignment between the loudspeaker and the model end face, a surface mirror was installed temporarily at the center of the latter, while a laser gun was placed normal to the speaker surface using the special adaptor bracket shown in Fig. 4. The alignment was accomplished by first leveling the top of the speaker enclosure using a carpenter's level, and then

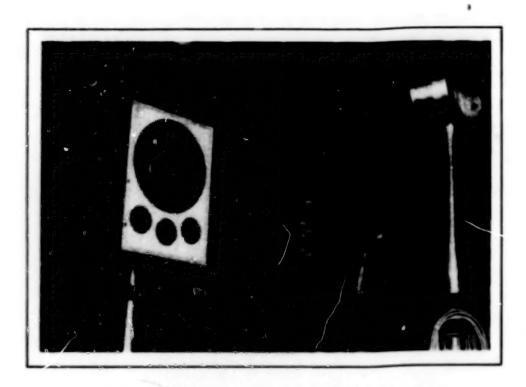
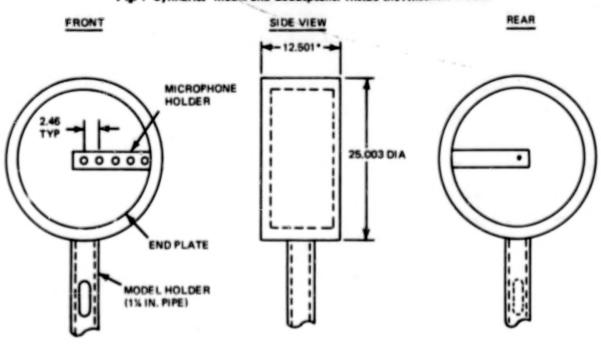


Fig. 1 Cylindrics' Model and Loudspeaker Inside the Anechoic Room



FOR THE L/D = 0.5 MODEL FOR THE L/D = 0.25 MODEL: 6.250 (ALL DIMENSIONS IN cm)

Fig. 2 Scattering Model

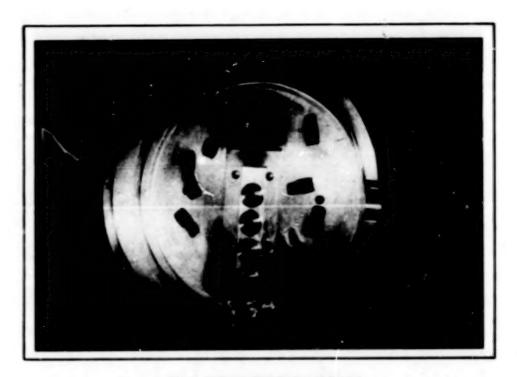


Fig. 3 Interior of Cylindrical Model

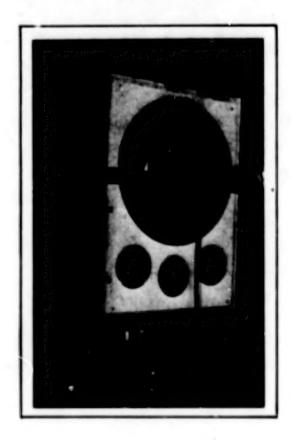


Fig. 4 Loudspeaker Adapter Bracket for Alignment with Laser Gun

turning the loudspeaker about a vertical axis and adjusting relative heights and the position of the model until the reflected laser beam almost coincided with the beam reaching the mirror.

Instrumentation

Acoustic surface pressures and phase angles were measured with six 0.635 cm diameter (pressure response) microphones in conjunction with commercially available right angle flexible connectors. The flexibility of the connectors made the microphone assembly less sensitive to mechanical vibration and made it possible to install five preamplifiers in a rather confined space. The microphone installation from inside the model is shown in Fig. 5. Microphones and their preamplifiers were electrically insulated from the model body, and thus from each other, by plastic preamplifier holders and nylon seals inside the flush mounting adaptors. Holes in the microphone holders were sized to ensure that the cartridges did not touch the microphone holder.

Sound pressure levels in the absence of the cylinder model were measured with a 1.3 cm diameter condenser microphone, which had been calibrated with a piston-phone calibrator.

Equipment used during the experiments to drive the speaker and to measure acoustic pressures, phase angles, and incidence angles is shown schematically in Fig. 6. Also shown is the computer that controlled the experiment. In addition to this instrumentation, anechoic room temperature and barometric pressure were also recorded and stored in the computer. These values were used to compute the speed of sound. The photograph in Fig. 7 shows some of the instrumentation used in this experiment.

Microphone Calibration

Microphones were calibrated for signal amplitude by means of a acoustic calibration piston phone at a SPL of 124 dB at 250 Hz. Phase response for each microphone was determined using an electrostatic actuator.

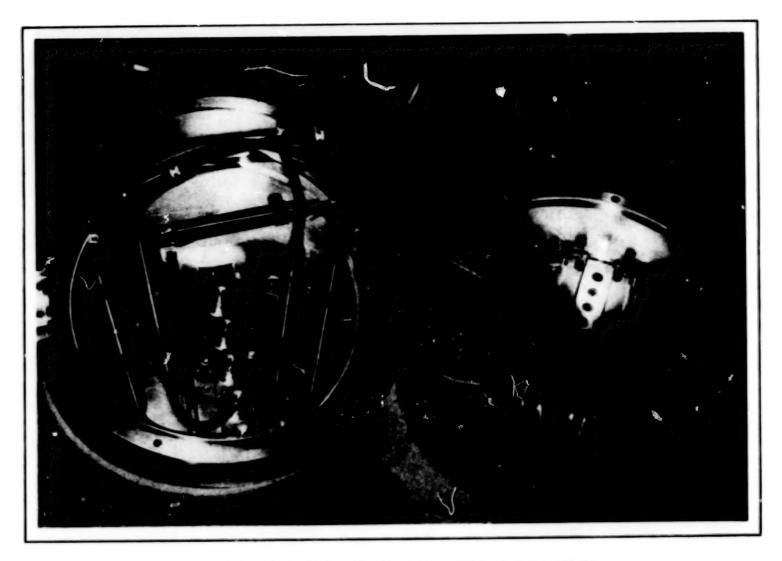


Fig. 5 Microphone and Preamplifier Installation Inside the Cylindrical Model

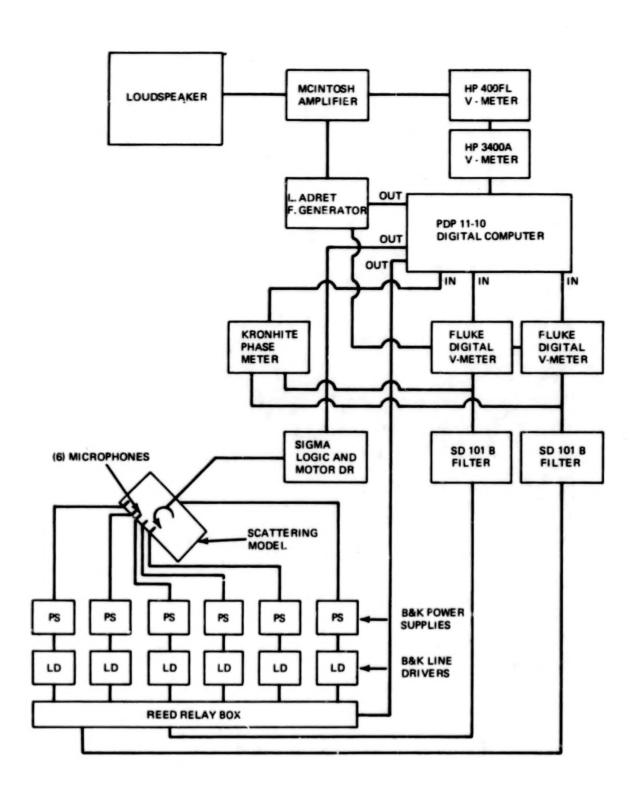


Fig. 6 Data Acquisition Circuit Schernatic

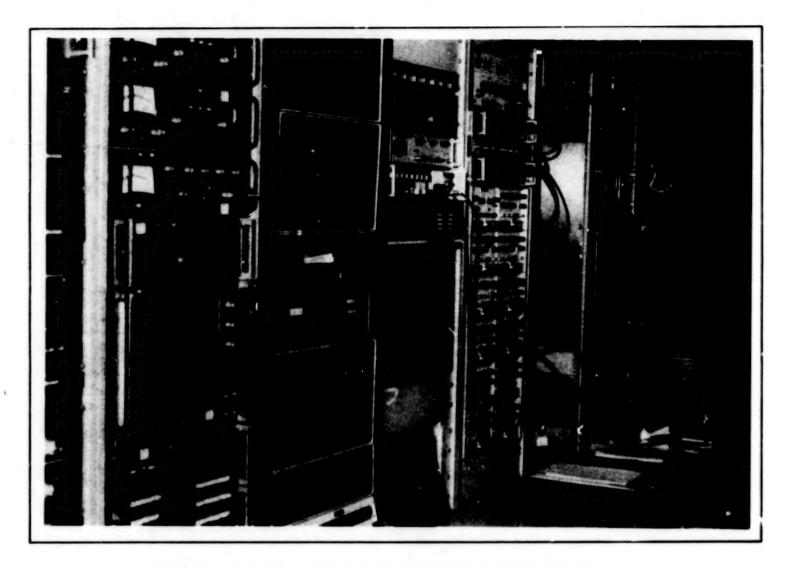


Fig. 7 Computer and Some of the Instrumentation Used in the Scattering Experiments

DISCUSSION OF RESULTS

Experimental results are presented in the form of three parameters: the surface pressure augmentation ratio π, relative phase angle, and the scattering parameter σ. Experimentally determined π and phase angle variations are compared with theoretical values provided by Thomas D. Norum of the Aeroacoustics Branch, NASA Langley Research Center. The methodology employed in the theoretical solution of the scattering problem for the particular problem of circular cylinders of finite length is described in Ref. 1. Numerical results became available after the initiation of the experimental program.

SURFACE PRESSURE AUGMENTATION

The surface pressure augmentation ratio π is defined as the absolute value of the ratio of the surface pressure existing at a given point on the surface of the cylinder body to the acoustic pressure that would exist at the same point in space in the absence of the cylinder body. It is, thus, a measure of the rearrangement in the acoustic field brought about by the introduction of the cylindrical body. The absolute value of the ratio is, of course, equal to the ratio of the rms values, which are the actual measurables. In the experimental procedure followed, each pressure measurement recorded is an averaged value over ten samples taken in quick succession, producing an averaging time of approximately 1.3 seconds. The reference acoustic pressures were obtained from SPL measurements at a position close to the model, with the model removed from the anechoic chamber. In computing π , corrections to the reference pressure were introduced for each model incidence angle to account for the difference in the loudspeaker to microphone distances for the microphones in the model surface.

Surface pressures were measured and π computed for 22 frequencies, ranging from 178 to 1850 Hz; in terms of the ka ratio, this covers a range from 0.407 to 4.232. Since, as mentioned previously, only one horizontal radius on one cylinder end face and the center point of the end plate on the opposite end were instrumented with microphones, the model was turned to eight different incidence positions and the model end face rotated 180° about its axis positions (from θ = 180° to θ = -135° in 45° steps) to

provide enough data for the front and back surfaces at three incidence angles: 180° , 135° and 90° . Figure 8 presents geometric definitions of the incidence angle θ and the end face nondimensional position x. Circular cylindrical models having L/D ratios of 0.5 and 0.25 were tested; in the latter case, the end face was not rotated 180° about its axis.

Results are presented for three ka values: 0.407, 2.288, and 4.232 (in terms of frequency: 178, 1000, and 1850 Hz). Results for intermediate ka values are tabulated in Appendix A.

Theoretical results are shown in Fig. 9 as solid lines for the front end face and as dash-dot lines for the back face. Triangles and circles stand for experimental measurements on the front and back end faces, respectively. The figure shows the experimental and theoretical pressure augmentation (π) distributions on the front and back of the L/D = 0.5 model end faces for ka = 0.407. As expected from Ref. 1, at this low ka value the effects of the model's presence are quite weak. The agreement between experimental and theoretical values is excellent.

The corresponding π distributions for the same cylindrical model at ka = 2.288 are shown in Fig. 10. Scattering effects are very clearly evident. At θ = 180°, the pressure at the center of the front face is about three times its free-field value. On the back face, which also exhibits strong scattering effects, one can see the beginning of the development of a diffraction ring pattern, which is well known in optics. The agreement between experimental and theoretical values is again, very good. Curves of π for the same model at ka = 4.232, shown in Fig. 11, attest to a very strongly scattering-dominated situation having a more elaborate fine structure. Diffraction rings on the rear face are fully developed. At $\theta = 135^{\circ}$, the peak value is located about half way on the radius further removed from the sound source. The agreement between theory and experiments is still quite good, but not as perfect as for the lower frequency cases. Slight disagreements can be noted at those positions where the theory predicts rather steep local peaks or valleys. As to the possible causes of these slight disagreements, it is important to realize that the experimental and theoretical representations of π differ in that each represents a different finite element of surface. The theoretical surface element is that of a cone, whereas the experimental element

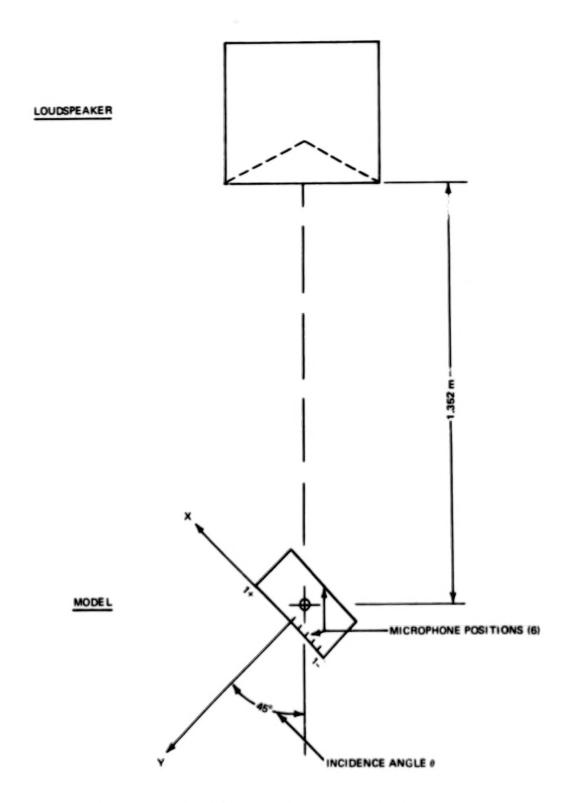


Fig. 8 Schematic Plan View of the Scattering Experiment Geometry

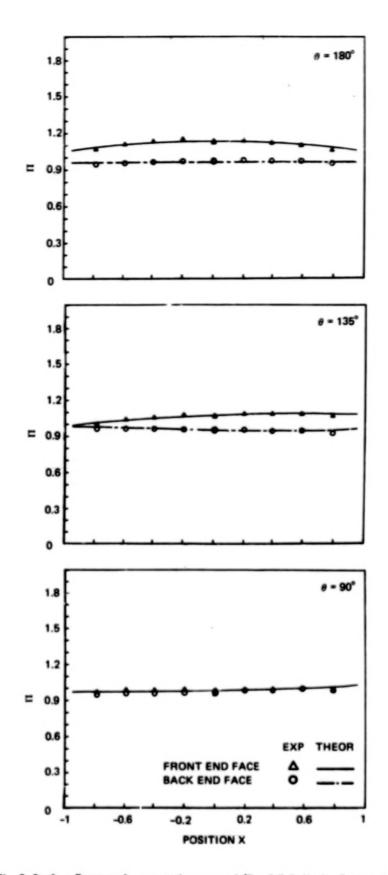


Fig. 9 Surface Pressure Augmentation on an L/D = 0.5 Cylinder Body at ka = 0.407

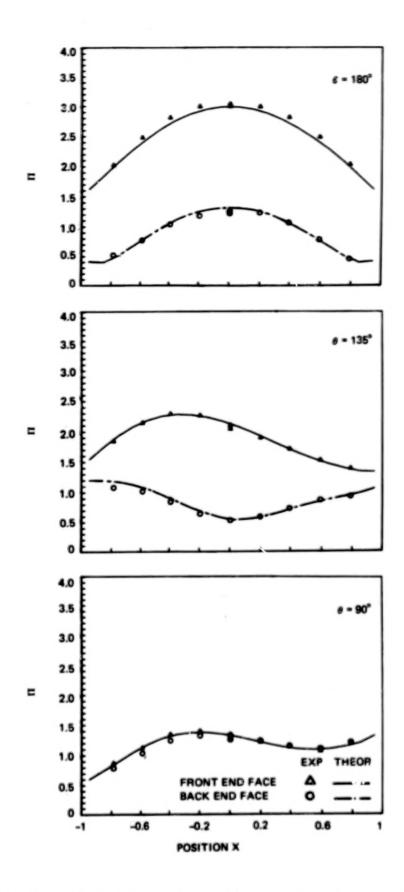


Fig. 10 Surface Pressure Augmentation on an L/D 5 Cylinder Body at ka = 2.288

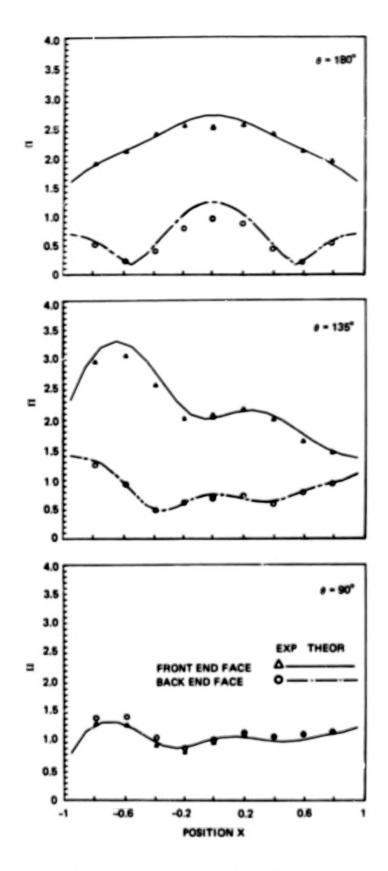


Fig. 11 Surface Pressure Augmentation on an L/D = 0.5 Cylinder Body at ka = 4.232

is that of a circular microphone. In both cases the elements are not small at high values of ka, and the difference between computed and measured distribution of π at high ka is expected. Furthermore, one should also keep in mind that whereas the theory assumed an ideal point source and a perfectly anechoic environment, in the real world no loudspeaker is a true point source at all frequencies and no room is perfectly anechoic. Sharp peaks or valleys in surface pressure would require near perfect local signal addition or cancellation; these are extremely difficult to duplicate in an experimental environment.

Figures 12-14 show comparisons of experimental pressure augmentation results with theory for a cylinder having half the length (L/D = 0.25) of that used to obtain data just presented. Trends are similar to those for the longer cylinder; however, there are noticeable differences in the corresponding π values. Generalizations on the effects on scattering brought about by halving the L/D ratio are hard to make. The effect seems to depend on the sound frequency and on the incidence angle. For instance, one can see by comparing Figs. 11 and 14 that at ka = 4.232 the shorter cylinder showed stronger scattering effects at θ = 180° and 90°, but at θ = 135° π values for the longer cylinder were larger.

Agreement between theory and experiment for L/D = 0.25 is generally very good, except for $\theta = 180^{\circ}$ at ka = 4.232 (Fig. 14). For some unknown reason, experimental data fell about 0.6 dB higher than the theoretical prediction at the center and about 0.8 dB higher towards the cylinder edge. It was not possible to repeat this particular configuration.

PHASE ANGLE VARIATIONS

Pressure phase variations were measured in conjunction with rms pressure measurements on the same two cylindrical models over the same range of test conditions. Phase angles relative to the microphone signal at the center of the model were measured with a phase meter. Results of the pressure phase measurements are presented in Figs. 15 through 21. As before, triangles and circles represent measured data on the model front and rear face, respectively, while the corresponding theoretical results are indicated as solid, or dash-dot lines. All phase angles are relative to

the phase at the center of the model front end face. Note that the plots showing phase variations corresponding to ka = 0.407 have an expanded scale to better display the modest phase differences at this low frequency. As expected, phase differences become more pronounced as the ka product increases. Compared to variations of pressure augmentation, curves of relative phase exhibit much less fine structure and have a more monotonic character. A comparison of corresponding phase variations for the L/D = 0.5 and 0.25 cylinder bodies leads to the following not surprising conclusions: First, the trends of phase variations are similar in both instances; secondly that at the θ = 90° incidence angle the variations are practically the same, as one would expect; and, thirdly, that at incidence angles other than 90°, the phase differences between the cylinder front and rear end faces are larger for the longer cylinder. This is also logical since the path the sound waves have to travel between corresponding points is longer for the longer cylinder. An attempt was made to collapse the experimental results into curves valid for both L/D ratios by including simple geometric considerations. This did not lead to a satisfactory result.

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The agreement between experimental phase data and theoretical predictions throughout the entire range of experimental variables is excellent. This is very rewarding since it represents a convincing verification both of theory and of the experimental appraoch, including such things as the suitability of the anechoic chamber, model design, and choice of loudspeaker and instrumentation.

SCATTERING PARAMETER

Whenever an acoustic pressure gradient at a point in space is measured with a pressure-difference sensor of finite dimensions, inaccuracies are introduced into the pressure gradient determination both by the scattering effects owing to the sensor's body and to the need for approximating a gradient from finite difference measurements. To assess the extent of these effects in a particular test configuration, it is expedient to introduce a scattering parameter σ . This parameter was first introduced in Ref. 2 and was used later extensively in Ref. 1. It is rederived here in Appendix B for the

particular case of spherical sound waves. It can be noted frow the derivation that in terms of decibels σ would equal zero if the measurement could yield the actual free field values.

Figure 21 shows a plot of σ in decibels vs. the ka product, as computed from experimental data for the L/D = 0.5 and 0.25 cylinder models. Incidence at θ = 180° and measurements from single pairs of microphones (front and back at x = 0) were used to prepare this graph. Two sets of experimental rata are shown for the L/D = 0.5 case; they correspond to the microphones on the model being either between x = 0 and x = 1, or between x = 0 and x = -1 (see Fig. 8), respectively. These data are identified by different symbols in order to display the degree of reproducibility. Also shown in the figure are the theoretical predictions for L/D = 0.5 and L/D = 0.25 from Ref. 1. In spite of the fact that in Ref. 1 plane acoustic waves had been assumed, the agreement with experimental values is satisfactory. It has been shown previously (Ref. 2) that the results are not much different if larger front and back areas are included. It is apparent from Fig. 21 that the L/D = 0.5 cylinder is better suited for PG microphone applications since its indicated acceptable upper frequency limit is higher.

The practical usefulness of scattering parameter plots of the type depicted in Fig. 21 becomes evident in microphone size selection, once the allowable σ value and the maximum operating frequency are fixed.

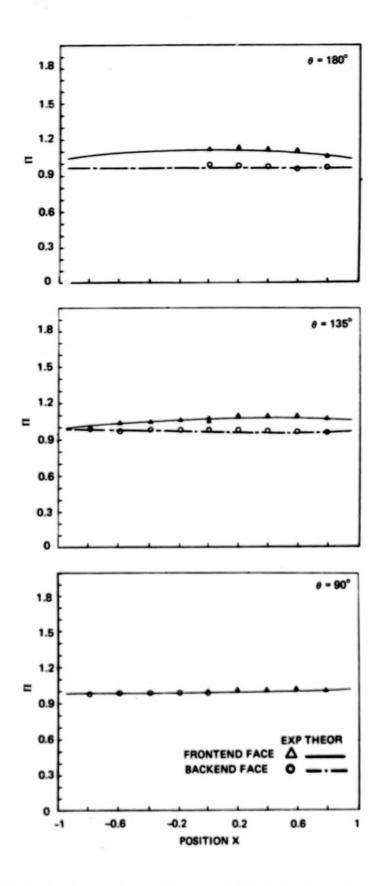


Fig. 12 Surface Pressure Augmentation on an L/D = 0.25 Cylinder Body at ka = 0.407

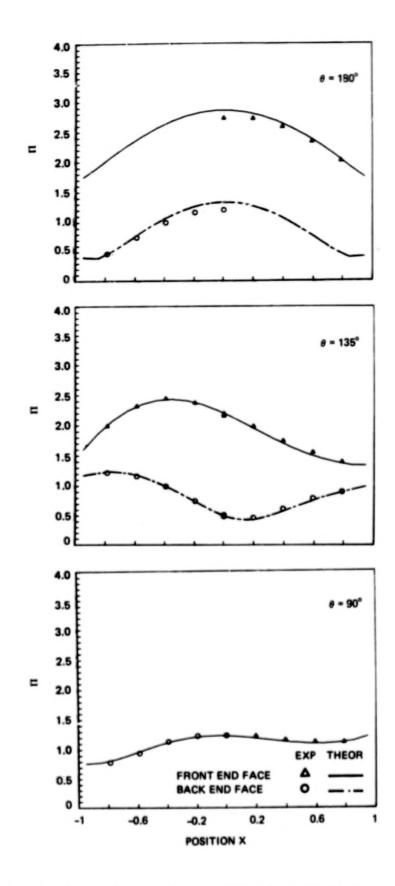


Fig. 13 Surface Pressure Augmentation on an L/D = 0.25 Cylinder Body at ka = 2.288

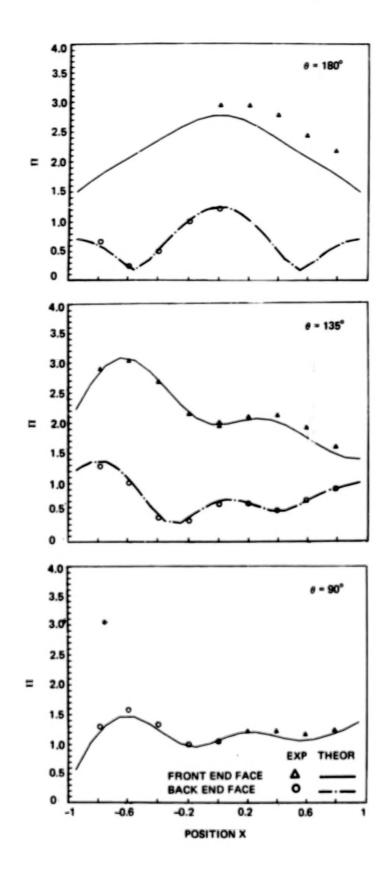


Fig. 14 Surface Pressure Augmentation on an L/D = 0.25 Cylinder Body at ka = 4.232

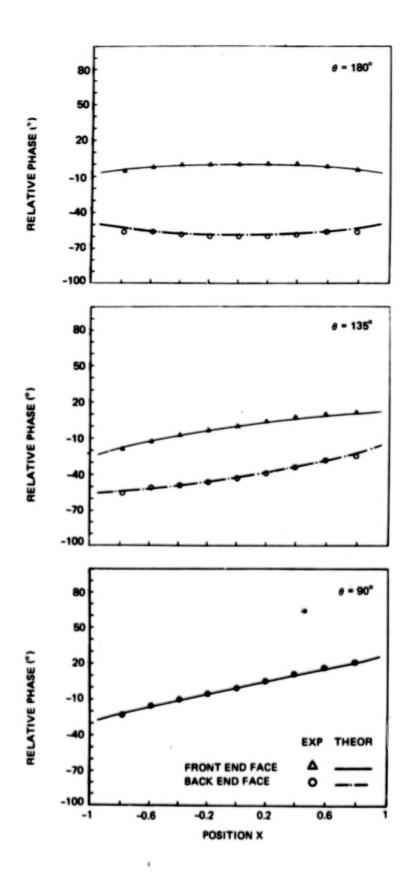


Fig. 15 Pressure Phase Variations on an L/D = 0.5 Cylinder Body at ka = 0.407

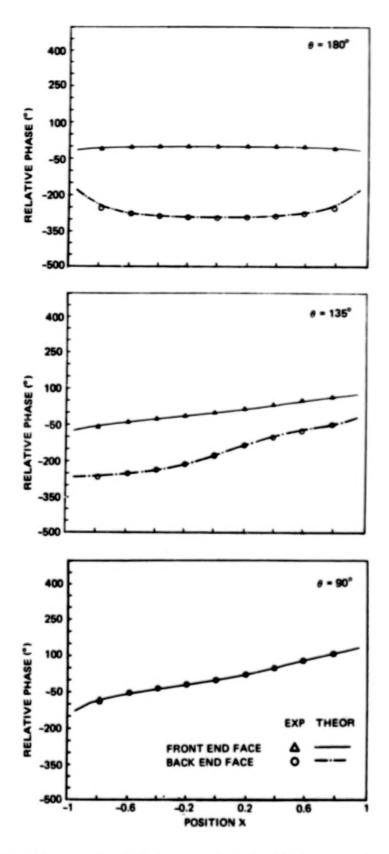


Fig. 16 Pressure Phase Variations on an L/D = 0.5 Cylinder Body at ka = 2.288

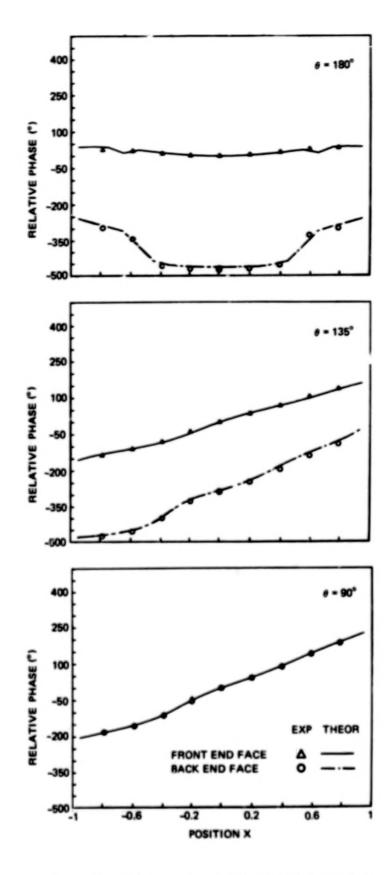


Fig. 17 Pressure Phase Variations on an L/D = 0.5 Cylinder Body at ka = 4.232

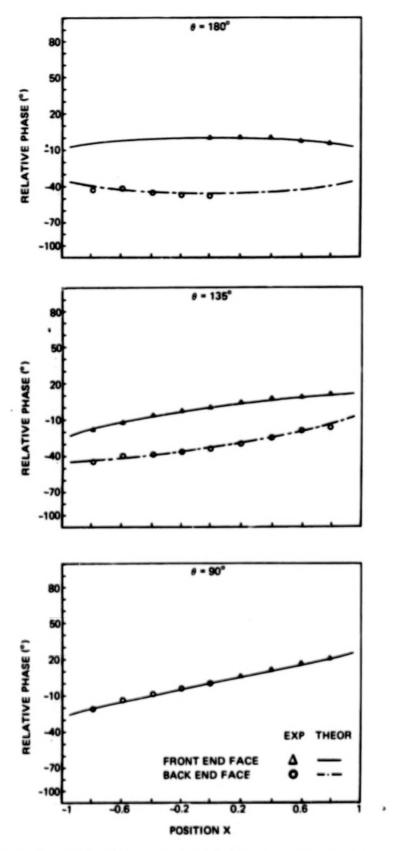


Fig. 18 Pressure Phase Variations on an L/D = 0.25 Cylinder Body at ka = 0.407

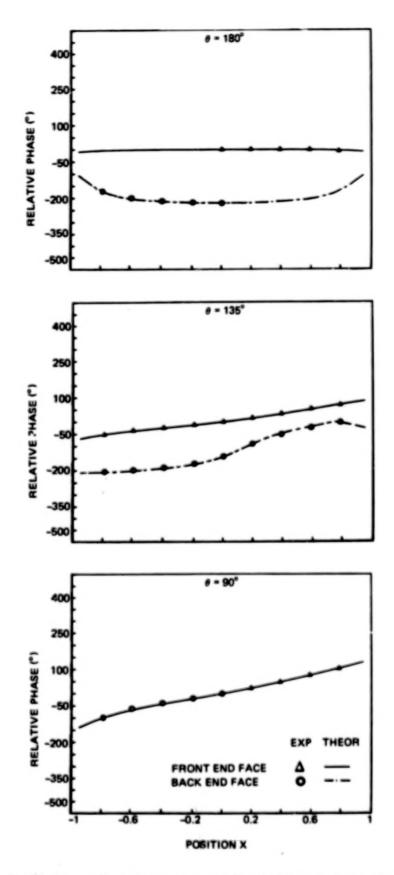


Fig. 19 Pressure Phase Variations on an L/D = 0.25 Cylinder Body at ka = 2.288

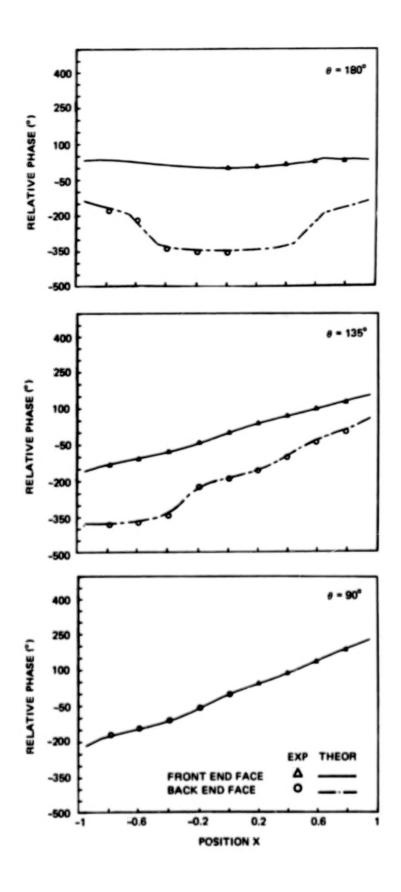


Fig. 20 Pressure Phase Variations on an L/D = 0.25 Cylinder Body at ka = 4.232

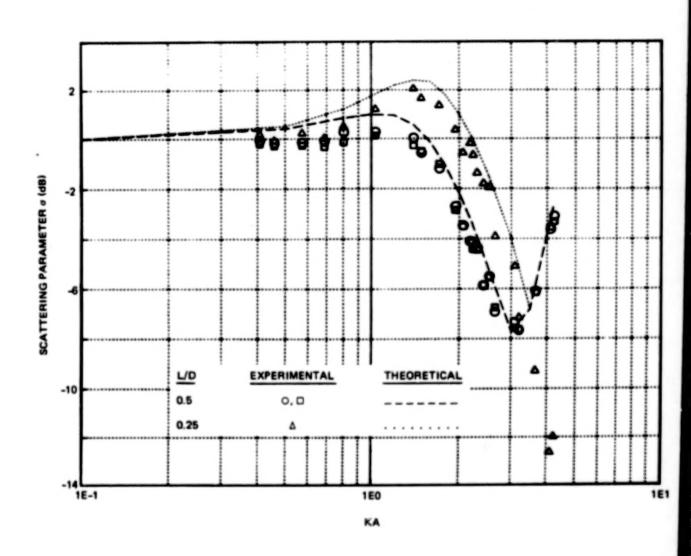


Fig. 21 Variations of the Scattering Parameter as a Function of ka for Circular Cylinders Having L/D of 0.5 and 0.25

CONCLUSIONS

Surface pressure augmentation and pressure phase changes trought about by acoustic scattering in a spherical sound field were experimentally determined for short circular cylinders having L/D ratios of 0.5 and 0.25. As expected, scattering effects were found to become more pronounced as the sound frequency was increased. Pressure augmentations in excess of three were measured.

Very good agreement was achieved between measured pressure augmentations and phase differences with theoretical values supplied by the Aeroacoustics Branch of the NASA Langley Research Center. This attests both to the suitability of the experimental approach and to the reliability of the theoretical model. Based on these results, scattering parameter variations were computed for both cylindrical bodies.

The scattering parameter curves are very useful in the preselection of proper geometries for pressure gradient microphones. It has been found that the longer cylinder model (L/D = 0.5) would possess a wider frequency operating range. This validates the findings arrived at theoretically in Ref. 1.

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APPENDIX A

TEST RESULTS

Measured pressure augmentation ratios and relative phase angles for 8 incidence angles and 22 frequencies between 178 to 1850 Hz are fully tabulated in this appendix. The tabulation is divided into three sections. Case 1 contains data for the L/D = 0.5 cylinder with the microphones 1-5 at the nondimensional positions X = 0, -0.1968, -0.3931, -0.5905, and -0.7873. Position 6 is always on the opposite cylinder end face at X = 0. Case 2 covers the same cylinder but for microphones rotated 180° about the cylinder axis, i.e., the X values are 0, + 0.1968, etc. Case 3 contains data for the L/D = 0.25 cylinder with the microphone positions as in Case 2. Pressure augmentation ratios are designated by PI, phase angles relative to position by PH.

CASE 1

THETA	FREO	PI(1)	PI (2)	PI (3)	PI (4)	PI (5)	PI (6)	PH(2)	PH(3)	PH(4)	PH(5)	PH(6)
190.000	178.000	1.136	1.150	1.134	1.110	1.070	.988	017	245	-2.368	-5.476	-59.607
180.000	200.000	1.129	1.149	1.134	1.105	1.062	1.052	.412	. 026	-3.347	-6.072	-65.670
180.000	250.000	1.218	1.244	1.236	1.208	1.166	1.057	. 339	576	-1.885	-7.270	-79.576
180.000	300.000	1.270	1.287	1.277	1.236	1.172	1.004	. 325	528	-2.808	-7.735	-99.630
180.000	350.000	1.419	1.423	1.404	1.348	1.272	1.019	.230	606	-2.808		-112.670
180.000	450.000	1.682	1.717	1.671	1.613	1.492	1.116	. 427	340	-3.224		-142.696
160.000	610.000	2.107	2.096	2.030	1.830	1.720	1.251	2.771	2.002	-1.457		-181.000
180.000	650.000	1.988	2.092	2.064	1.996	1.724	1.331	396	.623	-1.270		-190.437
180.000	750.000	2.335	2.302	2.236	2.051	1.836	1.349	.166	-1.985	-4.994		-222.440
180.600	850.000	2.472	2.504	2.310	2.159	1.896	1.338	479	-1.061	-4.196		-254.136
180.300	900.006	2.561	2.492	2.375	2.147	1.856	1.302	388	550	-3.937		-263.379 -280.973
180.000	950.000	2.897	2.722	2.534	2.314	2.013	1.371	286	611 952	-3.124 -3.577		-287.233
	:000.000	3.039	3.001	2.822	2.432	2.024	1.369	. 428	094	-2.531		-294.231
	1050.030	2.832	2.736	2.635	2.299	1.824	1.159	.506	.119	916		-303.512
	1100.000	3.252	3.244	3.046	2.569	2.170	1.272	. 419	.533	-1.247		-316.506
	1:50.000	2.911	2.884	2.682	2.312	1.951	1.035	. 490	1.262	.621		-323.762
	1350.000	3.175	3,169	2.949	2.526	2.848	1.137	.850	2.956	4.:09		-369.119
	1430.000	2.908	2.865	2.656	2.245	1.913	.939	.747	3.329	5.561		-379.621
	1600.030	2.731	2.319	2.636	2.258	1.973	1.029	1.784	5.971	11.384		-420.306
180.000	1300.090	2.547	2.631	2.539	2.197	1.956	1.156	3.897	11.035	18.747	24.427	-463.383
130.000	1350.000	2.473	2.490	2.356	2.943	1.961	1.065	2.257	9.856	20.554	24.701	-474.698
:35.000	178.390	4.675	1.030	1.057	1.940	1.001	.973	-3.294	-7.269	-12.524	-18.959	-42.716
135.000	200.000	1.935	1.649	1.031	1.020	.977	1.945	-3.837	-7.986	-15.163	-22.515	-46.6B6
135.000	256.000	1.126	1.128	1.101	1.053	1.015	1.015	-4.016	-9.414	-15.789	-25.384	-60.285
135.300	300.369	1.144	1.139	1.105	1.047	.998	.923	-5.178	-11.010	-18.509	-29.278	-73,235
135.900	350.000	1.240	1.234	1.168	1.101	1.029	.931	-5.644	-11.972	-20.710	-32.193	-80.459
135.000	450.000	1.446	1.437	1.346	1.256	1.114	.942	-5 . 488	-13.161	-22.732		-106.753
135.000	610.360	1.323	1.757	1.592	1.573	1.319	. 376	-9.198	-16.922	-27.672	-41.643	134.379
.35.000	659.000	1.732	1.794	1.723	1.534	1.301	.954	-7.394	-14.275	-23.385		-134.941
135.000	750.000	1.73:	: 926	1.874	1.715	1.476	.324	-0.314	-13.539	-29.581		-151.955
35.000	200.000	1.944	2.036	1.973	1.670	1.632	.677	-10.593	-21.253	-33.327 -36.148		-174.084
135.030	750.000	2.022	2.015	2.014	2.000	1.799	. 535	-11.644	-22.936 -23.525	-36.146		-181.171
135.036	979.030	2.039	2.210	2.237	2.079	1.875	.564	-12.342	-24.518	-38.026		-180.567
	1000.000	2.102	2.264	2.273	2.145	1.844	.577	-12.814	-25.130	-38.683		-184.396
	1953.030	1.934	2.145	2.194	2.094	1.776	. 486	-13.815	-27.408	-41.360		-181.307
	1100.000		2 . 364	2.446	2.335	1.950	. 525	- 4.510	-20.617	-43.874		-179.304
	1450.600	1.924	2.102	2.275	2.136	1.689	.528	-17.029	-31.671	-47.493		-187.019
	1350.000	2.009	2.445	2.718	2.670	2.275	.601	-23.200	-40.752	-58.739		-203.884
	1400.033	1.920	2.295	2.622	2.635	2.311	. 631	-26.348	-45.825	-64.046		-211.204
	1500.000	1.678	2.293	2.768	2.793	2.669	.695	-37.265	-61.953		-105.777	
	1300.000	2.102	2.179	2.765	2.957	3.132	.765	-41.117	-79.559	-105.514	-130.279	-276.790
135.000	1350.000	2.034	2.013	2.570	3.035	2.955	.743	-30.638	-90.546	-110.151	-136.370	-287.725

CASE 1

THETA	FREQ	PI(1)	Pi (2)	PI (3)	PI (4)	PI (5)	PI (6)	PH(2)	PH(3)	PH(4)	PH(5)	PH(6)
90.000	178.000	.984	.993	. 286	.988	.970	.999	-4.648	-9.505	-15.171	-22.511	. 334
90.000	200.000	. 940	. 965	.971	.997	.978	1.055	-4.989	-10.556	-16.515	-24.541	2.155
90.000	250.000	. 951	.954	. 946	.919	.912	1.014	-6.545	-13,900	-22.924	-32.473	-3.718
90.000	300.000	. 935	. 955	. 940	. 926	.922	.900	-7.993	-16.417	-26.154	-36.700	462
90.000	350.000	. 957	. 958	. 947	.930	. 925	.996	-9.864	-19.505	-30.577	-42.923	3.330
90.000	450.000	. 976	.958	.912	.900	.872	.954	-10.327	-25.549	-40.565	-57.846	-1.52.
90.000	610.000	.947	. 999	.949	.833	.794	1.145	-25.128	-39.122	-47.859	-03.827	-7 970
20.000	650.000	1.100	1.040	. 968	.887	.786	1.212	-15.965	-31.919	-53.552	-79.740	700
90.000	750.000	1.186	1.121	1.022	. 856	.720	1.377	-14.998	-31.788	-53.395	-62.654	.155
90.000	850.000	1.262	1.252	1.093	.924	:733	1.255	-16.009	-32.567	-52.626	-34.461	-2.303
90.000	900.000	1.286	1.300	1.210	1.032	.808	1.384	-17.363	-32.935	-53.365	-61.518	.011
90.000	950.000	1 . 265	1.277	1.175	.939	.734	1.323	-17.291	-34.551	-54.624	-82.595	-3.747
90.000	970.000	1.302	1.325	1.346	1.031	.773	1.367	-17.716	-34.460	-53.995	-80.276	-3.533
90.000		1.349	1.404	1.346	1.125	.872	1.434	-17.456	-34.536	-53.747	-80.597	-7.440
90.000		1.220	1.328	1.300	1.124	.771	1.149	-19.581	-37.201	-55.168	-39.375	1.190
90.000		1.232	1.361	1.339	1.137	. 827	1.203	-21.362	-39.381	-59.732	-85.349	.113
90.000		1.115	1.290	1 . 33:	1.135	.864	1.108	-23.646	-42.622	-61.192	-05.572	2.213
90.000		. 996	1.170	1.344	1.330	1.042	1.001	-38.417	-65.075	-68.195	-1:0.748	-2.758
90.000		.925	1.108	1.300	1.305	1.025	.961	-39.429	-67.065		-111.550	2.360
90.000		. 717	.947	1.209	1.434	1.253	.952	-56.139	-95.797		-148.923	-2.516
90.000		1.005	. 843	1.059	1.307	1.305	.999	-52,279				3.063
90.000		1.036	.839	.954	1.291	1.313	1.103	-45.688		-154.140		948
45.000	178.000	.965	. 979	.980	.987	.971	1.097	-3.210	-5.311	-7.866	-12.479	42.907
45.000	200.000	.965	. 991	.924	1.031	1.020	1.138	-2.571	-5.045 -8.525	-6.366 -14.401	-18.240	49.565 58.025
45.000	250.000		.913				1.147	-4.003 -4.368		-13.692	-18.015	72.77
45.000	300.000	.927	.976	1.008	1.012	1.007	1.128	-5.978	-8.616 -11.159	-16.251	-21.257	33.395
45.000	450.000	.909	.962	1.019	1.012	1.041	1.423	-7.333	-15.592	-21.390	-28.030	105.947
45.000	610.000	.785	1.013	1.029	.850	1.075	1.937	3.769	-4.368	-16.793	-20.998	152.659
45.000	650.000	.764	.947	1.047	1.158	1.097	1.925	-14.903	-21.296	-28.593	-36.630	144.345
45.000	750.000	.705	.848	.994	1.078	1.119	2.132	-17.367	-30.190	-40.053	-47.894	155.302
45.000	850.000	.614	.769	. 709	1.041	1.110	1.937	-24.057	-41.324	-53.376	-63.109	162.279
45.000	900.000	.628	.768	.908	1.029	1.091	2.036	-25.032	-42.959	-56.187	-66.229	166.279
45.000	950.000	.535	.701	. 893	1.053	1.148	2.075	-31.276	-50.664	-64.135	-74.938	173.613
45.000	970.000	.520	.671	. 868	1.039	1.140	2.135	-34.931	-56.701	-70.618	-63.398	171.375
45.000		.543	.664	. 856	1.030	1.075	2.196	-36.965	-60.147	-75.897	-90.809	167.747
45.000		.509	.636	.846	1.055	1.152	1.939	-39.430	-64.547	-79.792	-88.440	179.363
45.000		.541	.620	. 844	1.078	1.194	2.107	-45.031	-74.788		-104.745	179.833
45.000		.550	.600	. 790	.996	1.083	1.985	-43.433	-74.467		-107.435	186.183
45.000		. 575	. 498	.703	1.024	1.194	2.040	-54.270		-128.256	-141.704	203.917
45.000		.614	.478	.611	.919	1.108	1.983	-49.849		-133.847		208.763
45.000		.674	.455	.500	.881	1.165	1.976			-153.205		240 772
45.000		. 775	.596	. 442	.851	1.224	2.168			-163.784		278.154
45.000		.741	.571	. 439	.846	1.231	2.162			-169.376		288.846

THETA	FREQ	PI(1)	PI (2)	PI (3)	P1(4)	PI (5)	PI (6) *	PH(2)	PH(3)	PH(4)	PH(5)	PH(6)
.000	178.000	.978	.980	.975	.972	.953	1.155	.246	1.596	3.991	3.520	50.125
.000	200.000	1.011	1.019	1.021	1.029	1.004	1.181	1.228	2.354	6.421	5.570	57.228
. 000	250.000	.912	. 905	.904	.894	. 846	1.281	1.478	3.157	3.535	7.224	32.361
.000	300.000	. 985	1.014	1.003	. 980	.952	1.272	1.321	3.112	4.696	7.783	99.253
.000	350.000	1.026	1.040	1.030	. 998	.959	1.459	. 990	3.191	6.113	10.192	112.905
.000	450.000	1.086	1.092	1.092	1.051	.977	1.695	2.319	4.134	8.273	13.291	144.525
.000	610.000	1.331	1.286	1.160	.815	. 382	2.177	12.535	15.325	17.105	30.942	197.569
. 000	650.000	1.163	1.230	1.166	1.106	.901	2.302	1.990	9.351	16.229	27.235	200.970
.000	750.000	1.270	1.266	1.203	1.032	.840	2.509	2.399	5.656	13.629	27.713	222.763
.000	850.000	1.257	1.249	1.097	.914	.676	2.531	1.293	6.416	15.058	31.780	253.450
.000	900.000	1.214	1.193	1.077	.870	. 651	2.646	2.043	6.984	16.323	36.315	262.166
.000	950.000	1.256	1.226	1.079	.826	.577	2.931	1.931	6.278	15.955	36.911	280.935
.000	970.000	1.279	1.244	1.087	.817	.533	3.038	1.814	6.517	16.505	35.527	286.199
	1000.000	1.279	1.246	1.078	.793	.457	3.299	2.332	6.655	17.415	38.97B	291.936
	1050.000	1.150	1.109	.941	.679	.407	2.359	2.235	7.669	20 761	60.985	302.502
	1100.000	1.223	1.178	.979	.662	.399	3.337	2.370	3.639	21.966	65.577	316.239
	1150.000	1.091	1.048	. 961	. 569	. 333	3.074	2.151	8.540	23.521	72.062	323.551
	1350.000	1.060	.998	.749	.382	. 279	3.262	2.562	9.949	32.444	124.638	369.245
	1400.000	.976	.907	. 662	.312	.291	3.:12	2.456	10.566	38.014	134.599	378.300
	1600.000	.947	.821	. 524	. 178	.426	2.871	.622	13.568	77.394	163.239	420.469
	1800.000	1.063	.856	. 440	.168	. 524	2.793	3.897	16.502	124.499	177.295	463.251
	1850.000	.959	.872	.441	.211	.537	2.549	6.039	20.595	147.899	177.381	476.030
-45.000	178.000	.963	.959	.947	.950	.928	1.096	3.920	€.783	14.584	13.637	43.254
-45.000	200.000	. 997	.995	.990	1.005	.969	1.090	4.522	10.679	16.938	21.5:6	47.715
-45.000	250.000	. 856	.851	.854	. 845	. 342	1.248	7.588	14.564	20.852	30.045	59.546
-45.000	300.000	. 936	.947	.929	.901	.896	1.128	7.034	14.401	22.400	30.9:3	73.513
-45.000	350.000	.944	.938	.924	.906	.902	1.253	8.398	17.567	27.693	38.636	34.174
-45.000	450.000	.933	.913	.870	.630	.861	1.448	12.040	22.647	36.903	51.605	139.355
-45.000	610.000	.755	.833	.771	.776	.803	1.601	27.481	51.401	88.578	96.736	150.732
-45.000	650.000	.755	.722	.721	.721	.772	1.924	16.805	40.544	61.127	88.015	143.662
-45.000	750.000	.770	.700	.684	.735	.641	2.048	24.150	50.855	80.225	104.521	153.021
-45.000	850.000	.588	. 539	. 578	. 686	.794	1.982	33.181	68.355	96.383	119.568	169.268
-45.000	900.000	.602	.574	.628	.722	.811	1.996	33.176	66.337	94.555	118.164	169.659
-45.000	950.000	.537	.520	.616	.752	. 933	2.131	41.235		-254.478	125.936	179.913
-45.000	970.000	.516	.535	.661	. 808	.846	2.202	43.666	77.300	104.319	127.576	179.440
-45.000		.534	. 593	.732	. 871	.931	2.364	42.055	74.325	99.975	127.138	177.550
		.516	.572	.710	.842	.918	2.044	44.477	76.308	103.092	128.311	181.740
	1050.000			.627	.937	.963	2:201	42.203	69.852	94.188	119.577	178.923
-45.000		.512	.657			. 927	1.998	39.613	69.204	96.370	124.874	187.395
-45.000		.507	. 626	.753	.819	.779	2.084	32.835	58.940	89.294	130.787	201.051
-45.000		. 557	. 787	.871	.809			29.974	56.677	93.197	140.083	209.488
-45.000		.601	.770	.600	.715	.700	1.946	29.974		123.479	175.053	242.963
-45.000		. 662	.741	. 638	. 589	.802	1.971	36.248	65.866	152.582	193.438	276.527
-45.000		.747	. 637	. 570	. 787	.947	2.173		94.981	157.420	199.471	285.912
45.000	1850.000	.732	.738	.597	.790	.934	2.112	37.174	94.753	157.420	177.4/1	205.712

CASE 1

THETA	FREQ	PI(1)	PI (2)	PI (3)	PI (4)	PI (5)	PI (6)	PH(2)	PH(3)	PH(4)	PH(5)	PH(6)
-90.000	178.000	.976	.986	.987	1.001	.987	.993	5.772	11.736	17.037	21.639	.725
-90.000	200.000	. 998	1.012	1.019	1.039	1.017	.937	6.793	13.208	18.742	24.226	432
-90.000	250.000	. 882	.918	. 950	.962	.987	1.131	8.836	16.623	24.053	29.888	. 144
-90.000	300.000	. 932	. 964	. 984	. 993	1.014	. 900	9.602	18.394	27.093	34.528	1.395
-90.000	350.000	. 961	. 999	1.033	1.055	1.079	.930	10.588	20.670	30.253	38.516	1.669
-90.000	450.000	. 953	1.023	1.045	1.094	1.108	.951	13.679	24.992	35.939	46.689	4.916
-90.000	610.000	1.009	1.113	1.194	1.312	1.193	1.162	7.314	20.760	34.559	49.232	-3.934
-90.000	650.000	1.075	1.150	1.197	1.197	1.206	1.198	15.088	30.475	45.767	63.863	-3.176
-90.000	750.000	1.263	1.319	1.336	1.306	1.279	1.263	16.557	32.038	50.416		1.395
-90.000	850.000	1.200	1.228	1.164	1.127	1.076	1.305	17.387	38.359	59.925	93.219	930
-90.000	900.000	1.285	1.249	1.193	1.132	1.137	1.352	19.620	41.830	72.519	99.816	2.794
-90.000	950.000	1.219	1.136	1.127	1.076	1.067	1.335	21.037	46.077	74.616	104.653	1.767
-90.000	970.000	1.262	1.213	1.140	1.084	1.097	1.534	23.025	49.854	86.734	109.771	3.768
	1000.000	1.334	1.247	1.160	1.115	1.157	1.288	26.204	50.736	90.115	120.320	-3.504
	1050.000	1.214	1.121	1.046	1.052	1.13*	1.290	28.793	62.848	93.229	127.607	357
	1100.000	1.199	1.034	1.015	1.032	1.172	1.182	33.082	70.525	:05.511	1 32 . 528	3.791
	1150.000	1.075	.988	1.121	1.514	1.367	1.0.9	41.425	82 411	145 222	145.14	1.15
	1350.000	.692	. 764	1.125	1.200	1.219	. 243	45.387	82.326	115.735	47.08	-3 55
	1600.000	.920	1.111	1.168	1.121	1.112	.041	45.373	79.422	118.707	159.598	2.198
	1300.000	.962	1.039	1.021	1.009	1.135	1.035	40.948	65.952	137.525	179.473	1.475
	1850.000	1.053	1.172	1.102	1.141	1.182	1.069	40.735	86.55	139.496	184.63:	-1.250
-135.000	176.000	1.000	1.087	1.037	1.035	1.067	.971	3.304	7.132	2.421	11.900	42 21.
-135.000	200.000	1.073	1.102	1.108	1.105	1.331	.990	4.558	8. 03	10.283	12.550	47 672
-135.000	250.000	1.072	1.120	1.144	1.151	1.142	1.059	5.434	9.229	13.160	13.786	-55.732
-135.000	300.000	1.119	1.162	1.182	1.187	1.170	.913	6.146	11.122	15.291	6.807	-72.217
135.000	350.000	1.232	1.277	1.302	1.297	1.274	.918	6.708	11.922	15.822	18.154	-3:.726
-135.000	450.000	1.393	1.465	1.405	1.406	1.416	.9:3	7.766	13.491	17.930	21.185	103.329
-135.000	616.000	1.780	1.753	1.741	1.573	1.501	.681	0.147	16.666	22.985	29.763	-129.560
-135.000	650.000	1.704	1.730	1.738	1.677	1.549	.975	9.167	18.926	26.715	34.786	1.37.766
-135.000	750.000	1.961	1.914	1.627	1.671	1.524	.763	10.438	19.399	30.102	35.728	155.825
-135.000	850.000	1.873	1.612	1.640	1.512	1.358	.625	11.693	26.159	38.165	49.300	159.750
-135.000	900.000	1.877	1.774	1.648	1.438	1.357	. 687	13.538	27.929	42.217	55.129	164.382
-135.000	950.000	1.954	1.624	1.662	1.482	1.349	. 580	14.416	29.301	45.309	56.420	171.103
-135.000	970.000	2.010	1.861	1.636	1.503	1.341	. 571	14.938	30.752	47.72		-173.21
	1000.000	2.069	1.909	1.722	1.539	1.378	.641	16.161	33.434	51.375		-167.39
	1050.000	1.974	1.801	1.605	1.426	1.304	.548	17.132	35.727	\$5.573	72.309	134.585
	1100.000	2.052	1.846	1.652	1.492	1.417	. 554	19.498	40.738	62.768		-101.202
	1150.000	1.911	1.715	1.535	1.387	1.300	.571	20.250	43.210	66.465		-193.736
	1350.000	1.938	1.734	1.586	1.660	1.597	.589	31.170	61.925	86.233	104.261	-204.740
	1400.000	1.853	1.633	1.673	1.645	1.544	. 509	31.833	62.519	98.410	104.976	-207.624
	1600.000	1.881	1.956	2.015	1.885	1.606	.676	40.296	66.473	90.562	111.161	-239.778
	1800.000	2.050	2.146	2.025	1.778	1.513	.781	32.226	63.489	79.886	128.231	-276.272
-135.000	1850.000	2.060	2.129	1.982	1.762	1.440	.754	30.249	63.569	77.886	132,316	-230.61

THETA	FREQ	PI(1)	P1 (2)	P1 (3)	PI (4)	PI (5)	PI (6)	PH(2)	PH(3)	PH(4)	PH(5)	PH(6)
.33.000	179.000	1.124	1.138	1.120	1.190	1.061	.979	. 396	. 345	-1.962	-4.585	-59.203
106.000	200.505	1.1.21	1.146	1.146	1.132	1.100	1.938	.882	.365	-3.247	-4.980	-65,313
186.000	250.000	1.208	1.218	1.201	1.175	1.122	1.025	00B	324	-1.661	-6.625	-79.897
280.000	360.980	1.250	1.270	1.256	1.215	1.159	.981	.771	166	-2.235	-6.760	-97.952
190.000	330.000	1.370	1.443	1.404	1.356	1.205	1.913	.372	472	-2.280		-112.387
100.000	450.000	1.651	1.671	1.025	1.545	1.439	1.073	315	.036	-4.050		-141.555
180.100	613.800	2.020	2.071	2.040	1.854	1.712	1.221	1.507	1.577	559		-101.317
150.300	630.000 750.000	2.005	2.023	2.150	1.902	1.774	1.353	704	-2.562	-5.946		194.701
85.900	950.000	2.323	2.360	2.273	2.124	1.929	1.400	. 452	-1.616	-3.888		-221.811
		2.517			2.364	1.624	1.308	107	305	-3.137		-253.618
100.000	916.000	2.671	2.437	2.367	2.123	1.652	1.247	1.175	.583	-1.050		-263.522
.90.000	978.609	2.617	2.649	2.522	2.250	1.930	1.321	. 877	426	-1.490		-278.049
	1030.000	2.001	2.977	2.634	2.361	1.935	1.349	.555	.500	-1.738		-235.122
	1050.000	2.772	2.764	2.000	2.497	1.860	1.354	.805	.686	-1.129		-293.952
	1:50.000	3.212	3.204	2.997	2.613	2.036	1.169	.564	.794	327		-301.732
184.000		2.901	2.929	2.719	2.316	863	1.109	566	1.762	531		-315.253
89 000		3.115	3.100	2.877	2.404	2.061	1.120	1.344	4.199	5.930		-322.959 -367.541
130.300		2.923	1.875	2.659	2.263	1.939	1.100	1.531	4.497	7.639		-367.541
	1030.300	2.731	2.746	2.562	2.203	1.630	.991	3.187	8.194	14.331		416.700
	1300.000	2.623	2.403	2.413	2.009	1.842	1.051	5.367	13.670	23.414		-460.425
180.036		2.451	2.498	2.338	2.072	1.90;	1.055	5.357	15.456	27.735		-470.779
135.000	178.000	1.068	1.039	1.007	1.008	1.072	.971	3.991	7.582	9.931	11.390	-42.587
135.000	200.040	:.039	1.072	1.036	1.037	1.081	1.932	5.209	9.151	10.381	13.039	-45.772
135.030	250.300	1.124	1.139	1.173	1.1"9	1.159	.979	4.501	0.345	12.750	13.711	-60.228
125.000	300.000	1.138	1.179	1.170	1.174	1.176	.910	0.101	10.643	14.649	16.599	-71.533
:35.030	350.000	1.209	1.247	1.297	1.275	1.279	. 931	6.455	11.972	15.426	18.246	-08.490
135.000	450.000	422	1.494	: . 4.*4	1.142	1.420	.926	5.835	13.194	15.798		-104.583
35.000	613.030	1.795	1.775	1.740	1.640	1.535	. 962	0.328	16.149	22.344		-129.947
135.000	050.000	710	1.728	1.738	1.620	1.559	. 914	B1179	15.307	23.917		-138.825
135.000	750.000	1.990	1.896	1.792	1.673	1.543	. 345	10.050	19.345	29.441	30.042	-149.501
135.00.	850.000	1.037	1.775	1.647	1.436	1.353	.633	11.006	25.246	30.421		-170.398
135.000	700.000	1.916	1.600	1.675	1.514	1.372	.627	14.071	28.279	42.936	54.953	-167.342
135.936	950.000	1.963	1.851	1.632	1.476	1.331	.598	14.681	29.645	45.670	59.511	-177.698
135.000	970.000	2.000	1.669	1.638	1.477	1.350	.573	14.826	30.817	47.905	63.098	-178.798
135.000		2.043	1.697	1.710	1.523	1.335	.514	15.977	33.350	51.091		-196.555
135.000		1.919	1.770	1.601	1.431	1.307	.515	17.259	35.673	55.567		-177.496
	1100.000	2.096	1.890	1.680	1.503	1.400	. 529	10.392	39.112	60.982		-177.629
135.000		1.909	1.719	1.547	1.376	1.345	. 543	20.652	43.867	67.126		-102.275
135.030		1.935	1.734	1.662	1.056	1.604	. 682	30.600	61.271	87.554	105.051	
135.000		1.611	1.660	1.634	1.676	1.567	.611	31.386	61 677	96.301	103.506	
135.000		1.900	2.032	2.001	1.670	1.572	. 681	36.687	64.520	98.990	109.525	
135.000		2.124	2.199	2.039	1.679	1.437	.743	35.325	66.103	97.654	1.79.480	
135.000	1850.000	2.071	2.165	1.997	1.634	1.446	. 761	34.905	67.537	104.773	137.417	-283.544

THETA	FREQ	PI(1)	P1(2)	PI (3)	P1 (4)	PI (5)	PI (6)	PH(2)	PH(3)	PH(4)	PH(5)	
90.000	178.000	.983	.993	.996	1.094	. 989	.987	5.577	11.116	16.483	21.217	177
90.000	200.000	.955	.961	.966	.983	.971	1.040	6.862	13.694	19.339	25.008	1.473
99.000	250.000	.951	.991	1.014	1.027	1.051	.977	7.638	15.123	22.520	28.448	-3.255
98.880	300.000	.949	.979	. 578	1.009	1.024	.904	9.451	17.993	26.698	34.065	. 872
90.000	356.000	.941	. 783	1.0:6	1.037	1.059	.945	11.104	28.974	30.515	39.435	3.249
90.000	450.000	. 959	1.051	1.650	1.083	1.136	.949	12.337	24.986	34.429	45.047	. 834
90.000	610.000	1.062	1.133	1.175	1.293	1.248	1.131	7.335	20.204	33.135	47.645	-6.514
76.000	550.000	1.070	1.161	1.200	1.198	1.215	1.162	15.001	28.813	44.358	59.124	1.553
20.035	750.000	1.156	1.259	1.258	1.247	1.229	1.433	16.364	31.988	50.037	68.100	051
90.680	950.000	1.224	1.218	1.132	1.121	1.074	1.216	17.763	37.681	59.874	82.711	-1.220
90.000	986.888	1.302	1.265	1.204	1.152	1.151	1.379	29.733	42.580	60.405	93.812	.142
90.900	950.000	1.226	1.211	1.151	1.111	1.074	1.320	22.142	46.415	73.353	99.936	-3.427
50.000	970.000	1.243	1.234	1.139	1.138	1.093	1.352	21.558	46.284	73.922	104.440	-4.759
70.080	1909.090	1.293	1.223	1.142	1.075	1.194	1.226	22.351	48.266	77.707	106.850	-10.666
	1950.000	1.206	1.111	1.034	1.040	1.133	2.599	25.906	56.774	90.836	110.170	2.492
	1130.000	1.213	1.123	1.000	1.876	1.192	1.229	28.453	61.918	95.121	125.385	.100
	1150.000	1.110	.974	.970	1.065	1.212	1.137	31.742	70.431	107.432	135.738	1.425
	:350.000	.938	4.927	1.172	1.250	1.267	1.021	44.382	79.725	111.189	142.008	-2.110
	1 300 . 0 30	. 862	.915	1.001	1.159	1.201	. 936	44.538	82.873	116.918	147.010	-2.130
	1603.000	. 935	1.179	1.221	1.152	1.124	. 930	46.433	78.083	116.469	156.838	833
	1399.096	. 936	1.068	1.039	4 - 932	1.146	. 978	43.943	89.850	142.535	183.106	6.879
	1350.000	. 999	1.127	1.071	1.152	1.136	1.126	42.919	89.645	142.498	185.955	2.775
45.000	176.850	.962	.9 >2	.952	. 949	. 935	1.000	3.908	8.896	14.519	18.024	42.598
45.000	200.000	.977	.959	. 949	.945	.914	1.124	4.419	9.554	16.118	19.998	48.478
45.000	250.000	.394	.714	.919	.912	. 929	1.110	6.708	13.709	20.447	28.728	58.381
45.600	300.000	.929	.934	.931	.705	-966	1.120	7.065	14.291	22.384	31.149	71.817
45.040	350.000	. 934	.933	. 726	.984	. 890	1.264	9.408	17.181	27.178	30.128	83.965
45.000	450.000	. 705	. 706	.052	.853	. 958	1.412	12.052	22.667	35.938	52.488	107.221
	616.030	. 682	.769	.782	.602	.820	1.805	22.205	43.352	75.477	93.418	143.495
45.000	656.900 750.000	.761	. 737	.717	. 696	707	1.991	16.718	34.370	60.133	81.953	141.579
45.000	859.000	.609	.535	.612	.556	.759	1.924	24.550 32.361	52.763 65.779	94.174	107.491	155.118
45.000	900.000	.629	.595	. 653	.751	. 845	2.021	33.754	65.364	93.732	117.012	166.981
45.360	750.000	.552	.531	.625	.764	. 861	2.048	41.659	76.451	105.264	125.793	175.639
45.080	970.000	.522	543	.662	.610	.689	2.100	44.039	77.565	104.510	123.367	174.325
	1000.000	. 473	. Se 1	.692	.948	-836	1.935	43.213	76.119	101.722	128.198	170.749
	1050.000	.504	.575	.717	. 330	.915	1.987	43.131	74.135		-119.660	182.563
	1100.000	.515	.649	. 816	.920	.964	2.130	43.312	72.196	96.931	122.833	183.118
	1150.000	.514	. 669	.895	. 880	. 880	2.026	30.501	66.949	93.256	120.768	185.282
	1350.000	.514	. 753	.817	.768	.761	2.063	33.479	60.436	93.95	137.325	203.395
	1400.000	.601	.769	.605	.732	.717	1.761	26.878	53.051	98.64	134.193	203.071
	1600.000	. 600	.722	. 630	. 532	.787	1.989	28.653	65.770	124.311	175.421	239.750
	1800.000	.777	.711	.620	.746	.983	2.137	36.683	93.876	156.222	193.594	280.669
	1350.000	.758	.729	. 639	.794	.913	2.171	38.928	90.633	146.555	194.712	284.519

THETA	FREO	PI(1)	P1 (2)	PI (3)	P1(4)	PI (5)	PI (6)	PH(2)	PH(3)	PH(4)	PH(5)	PH(6)
.000	179.000	.968	.972	.966	.957	.946	1.141	. 360	1.421	4.367	4.243	59.863
.000	200.000	1.013	1.004	1.002	. 989	.969	1.167	. 158	1.520	4.349	2.931	66.191
. 000	250.000	. 897	.928	. 926	.923	.908	1.252	2.244	3.942	4.901	10.211	83.926
.000	300.000	.972	.998	.994	. 962	.929	1.266	1-130	2.935	4.971	7.846	98.238
.000	350.000	1.007	1.024	1.024	.995	.955	1.430	1.386	3.398	5.625	9.343	113.462
.000	450.000	1.075	1.093	1.075	1.048	.957	1.685	2.464	2.603	7.524	12.685	145.208
.000	610.000	1.169	1.254	1.205	. 938	. 863	2.140	10.236	16.035	23.279	34.376	196.360
.000	450.000	1.137	1.173	1.195	1.060	. 949	2.246	.783	2.934	8.915	22.112	197.913
.000	750.000	1.284	1.277	1.101	1.006	.601	2.592	. 935	3.390	10.198	22.315	220.205
.000	850.000	1.237	1.209	1.071	.093	.671	2.467	1.222	6.331	14.549	31.263	252.608
.000	700.000	1.215	1.102	1.045	.050	.622	2.656	2.234	7.122	16.910	36.912	263.420
.000	950.000	1.230	1.226	1.009	. 846	.550	2.790	2.416	7.239	17.359	37.967	279.862
.000	978.000	1.252	1.236	1.071	.834	.509	2.995	1.631	4.253	16.464	40.704	284.630
.000	1000.000	1.238	1.200	1.051	.780	.521	3.869	1.797	6.507	16.826	46.389	208,430
.000	1050.000	1.143	1.116	.962	. 696	.494	2.872	1.777	4.554	17.751	40.572	301.784
.000	1100.000	1.227	1.198	1.002	.692	.392	3.318	1.624	6.436	18.343	61.652	314.983
	1150.000	1.126	1.023	. 872	.548	. 343	3.179	1.633	7.509	22.017	71.099	322.989
	1350.000	1.035	.978	.723	. 345	.293	3.186	3.311	11.065	35.944	125.405	368.410
.000	1.400.000	.969	.920	.677	. 321	.284	3.171	1.019	10.022	37.386	133.343	366.982
. 839	1600.000	.911	.776	. 489	.177	. 432	2.968	.798	14.861	63.185	162.497	418.329
.000	1.000.000	1.045	.950	. 5 +6	.173	.500	2.711	2.304	13.946	186.002	172.320	468.883
.200	1850.000	. 949	.792	. 406	.231	.516	2.547	-1.249	11.062	126.664	173.411	468.694
-45,000	178.000	. 952	.963	. 964	.964	.962	1.083	-3.300	-5.705	-7.844	-12.616	43.169
45.000	200.000	. 995	1.060	1.012	1.015	1.006	1.077	-3.100	-4.258	-8.245	-13.647	47.897
-45.000	250.000	.837	.971	.070	. 677	.078	1.217	-3.682	-7.894	-13.245	-15.099	60.305
-45.000	300.000	.919	. 944	.977	.972	. 966	1.123	-4.639	-9.878	-13.605	-17.714	72.361
-45.000	350.000	.917	.957	. 984	. 989	.985	1.234	-5.787	-11.050	-16.464	-21.733	83.340
-45.000	450.000	.922	.995	1.015	1.073	1.042	1.447	-6.415	-16.107	-22.066	-27.623	107.001
-45.000	610.030	.673	.945	1.073	1.000	1.060	1.830	-9.672	-17.005	-22.373	-32.600	139.419.
-45.005	650.000	.769	.891	1.037	1.078	1.128	1.895	-16.044	-28.008	-36.751	-41.691	139.463
45.000	750.000	. 792	.925	1.044	1.110	1.149	2.995	-17.126	-31.131	-41.969	-50.517	149.782
-45.000	850.000	. 595	.735	.081	1.007	1.074	1.947	-24.259	-39.743	-51.794	-61.466	166.108
-45.000	760.000	. 620	.740	. 889	1.001	1.056	2.017	-25.210	-43.151	-57.101	-67.050	167.218
-45.000	950.000	. 531	.701	. 894	1.055	1.121	2.114	-38.512	-49.441	-62.786	-73.698	177.300
-45.000	970.000	. 522	. 6-6-6	. 676	1.040	1.093	2.201	-35.121	-54.004	-70.145	-82.041	174.018
	1000.000	. 534	. 646	. 647	1.021	1.083	2.250	-34.926	-59.267	-74.855	-62.535	172.150
	1050.000	.520	.644	. 640	1.031	1.122	2.020	-37.986	-63.132	-79.128	-90.521	177.381
	1100.000	. 539	. 605	. 826	1.000	1.100	2.205	-45.156	-75.840	-94.819	-105.891	175.000
	1150.000	.540	. 606	. 795	.977	1.000	2.022	-41.652	-72.849	-93.725	-106.501	184.997
	1350.000	.509	.501	.714	1.034	1.200	2.072	-56.856	-104.876	-129.786	-143.633	198.329
	1400.000	. 625	. 483	.627	.926	1.102	1.895	-49,459	-102.231	-131.192	-147.005	210.175
	1600.000	.674	.412	.504	.914	1.194	1.963		-118.857	-153.967		239.054
	1000.000	.707	.587	. 399	. 927	1.175	2.127		-104.870		-198.967	271.964
45.000	1050.000	.672	.624	.494	.932	1.253	2.159	-41.033	-109.780	-167.130	-171.281	282.548

THETA	FREQ	PI (1)	PI (2)	PI (3)	PI (4)	PI (5)	PI (6)	PH(2)	PH(3)	PH(4)	PH(5)	PH(6)
-90.000	178.000	.963	.966	.959	.960	.949	.982	-4.866	-9.714	-15.:81	-22.711	. 750
-90.000	200.000	.991	.998	1.002	1.014	.991	.979	-4.753	-9.364	-16:467	-24.227	712
-90.000	250.000	.860	. 873	. 862	.841	.846	1.105	-6.842	-14.928	-24.034	-32.385	.294
-90.000	300.000	.923	.940	.940	.919	.910	.882	-8.565	-16.856	-25.882	-36.385	. 613
-90.000	350.000	. 943	.940	.932	.989	.875	. 930	-9.226	-19.247	-30.519	-43.119	. 384
-90.000	450.000	. 950	.965	.330	.988	.884	.942	-12.211	-26.381	-42.499	-56.544	6.226
-90.000	610.000	1.111	1.002	.945	.907	.811	1.141	-20.014	-36.625	-53.449	-79.530	3.993
-90.000	650.000	1.078	1.020	.954	.850	.767	1.170	-15.356	-31.679	-53.707	-79.481	-2.905
-90.000	750.000	1.290	1.235	1.127	. 959	.826	1.271	-14.996	-32.271	-53.063	-81.606	-1.150 .942
-90.000	850.000	1.186	1.146	1.638	.860	.681	1.273	-16.238	-32.413	-53.601	-84.222	.148
-90.000	900.000	1.307	1.309	1.220	1.019	.795	1.385	-15.513	-31.793	-51.130	-78.562	2.799
-90.000	958.000	1.228	1.243	1.164	.964	.745	1.324	-16.937	-34.190	-54.288	-91.772	2.350
-90.000	970.000	1.256	1.290	1.203	.995	.793	1.391	-16.857	-34.547	-54.277	-91.155	-2.293
	1000.000	1.265	1.330	1.250	1.040	.790	1.472	-17.933	-34.694	-52.750	-88.809	-2.987
	1050.000	1.223	1.324	1.300	1.121	.606	1.257	-19.873	-37.187	-56.2J2	-84.079	-1.441
	1100.000	1.230	1.363	1.362	1.179	.841	1.289	-21.135	-30.327	-58.246	-84.784	-3.140
	1150.000	1.131	1.300	1.354	1.208	.891	1.208	-24.050	-42.630	-61.537 -87.290	-86.493	-1.578
	1350.000	. 994	1.186	1.364	1.341	1.031	1.004	-38.639	-64.874		-109.447	61
	1400.000	. 935	1.095	1.238	1.289	1.013	.925	-38 769	-66.567		-111.098	2.634
	1600.000	.914	.930	1.203	1.373	1.216	.937	-53.223	-92.524	-121.458 -145.395		-2.755
	1900.000	. 693	. 633	1.014	1.354	1.198	1.025	-53.192		-156.729		-3.431
	1350.000	1.047	.910	1.937	1.433	1.413	1.987	-54.233		-12.587	-19.376	-42 837
-135.000	178.000	1.649	1.048	1.026	1.008	.973	.958	-3.334 -3.965	-5.864 -7.415	-14.626	-20.594	-47.555
-135.000	200.000	1.065	1.074	1.963	1.031	1.614	1.038	-4.615	-9.800	-10.170	-25.744	-56.548
135.000	250.000	1.055	1.049	1.015	.967	.929			-11.417	-19.102	-29.417	-72.434
-135.000	300.060	1.113	1.098	1.069	1.015	. 969	. 392	-5.102	-11.791	-19.697	-30.659	-82.432
-135.000	350.000	1.218	1.202	1.164	1.093	1.013	.913	-5.168 -6.782	-13.037	-25.367	-37.570	-103.467
-135.000	450.000	1.395	1.375	1.265	1.183	1.278		-5.911	-13.313	-25.170		-1.29.286
-135.000	610.000	1.765	1.734	1.720	1.495	1.308	.937	-8.943	-17.425	-28.245		-140.784
-135.000	650.000	1.691	1.706	2.014	1.844	1.606	.788	-3.520	-18.840	-29.797		-157.718
-135.000	750.000		2.055	1.877	1.737	1.515	.614	-10.637	-20.603	-33.174		-162.489
-135.000	850.000	1.061	1.910		1.864	1.650	.665	-10.561	-21.576	-34.502	-50.841	-161.939
-135.000	900.000	1.915	2.003	2.045	1.934	1.723	.604	-10.053	-22.649	-35.658	-53.712	-171.437
-135.000	950.000	1.946	2.084		2.022	1.787	.593	-11.742	-23.583	-36.762		-170.703
-135.000	970.000	1.986	2.151	2.167	2.049	1.773	.668	-12.307	-24.831	-38.575		-176.309
-135.000		1.969	2.153	2.199	2.077	1.756	.536	-13.439	-26.729	-40.326		-185.315
					2.323	1.967	.554	-14.997	-29.725	-43.908		-183.589
-135.000		1.978	2.331	2.429	2.210	1.891	.577	-16.407	-30.975	47.939		-185.742
-135.000 -135.000		1.995	2.391	2.654	2.643	2.204	.571	-24.978	-43.055	-60.942		-208.640
-135.000				2.568	2.578	2.253	.591	-24.652	-44.154	-62.679		-205.727
		1.930	2.266		2.793	2.618	.671	-37.029	-61.854	-83.224		
-135.000		1.052	2.223	2.696			.765	-41.656	-77.616	-107.606	-128.797	
-135.000		1.909	2.123	2.687	2.957	2.935					-139.302	
-135.000	7620.000	1.973	2.073	2.672	3.055	3.079	.754	-45.386	-62.661	-116.732	-134.305	-207.173

THETA	FREQ	PI(1)	PI (2)	PI (3)	PI (4)	PI (5)	PI (6)	PH(2)	PH(3)	PH(4)	PH(5)	PH(6)
160.006	173.006	1.122	1.138	1.124	1.114	1.065	1.001	. 033	162	-3.184	-5.010	-47.392
150 000	200.600	1.122	1.147	1.:43	1.1.20	1.185	1.042	. 304	168	-4.799	-5.883	-51.279
190.000	256.000	1.179	1.187	1.180	1.145	1.114	.977	198	- 393	-1.569	-7.171	-67.077
180.700	300.000	1.226	1.237	1.227	1.176	1.122	1.054	021	847	-3.077	-B.346	-75.654
100.003	350.000	1.354	1.307	1.336	1.307	1.235	1.073	. 431	614	-2.914	-7.576	-89.647
189.000	450.600	1.649	1.045	1.639	1.516	1.405	1.154	-1.135	-2.960	-4.461		-115.141
.03.690	c19.300	2.239	2.207	2.134	2.057	1.824	1.393	-1.259	-1.479	-4.520		-151.767
1:0.000	5-0.000	2.238	2.276	2.206	2.023	1.824	1.345	. 492	.027	-2.301		-159.334
130.009	56.300	2 578	2.593	2.499	2.323	2.111	1.332	.013	- 039	-2.439		-173.619
130,000	850.003	532	2.6:3	2.477	2.275	2.040	1.348	1.353	.753	342		-193.985
61.016	963.000	2.519	2.522	2.414	2.193	1.967	1.275	650	.763	859		-199.468
1 16 - 3 - 6	950.000	2,6%3	2 600	2.732	2.162	2.149	1.405	.305	. 499	681		-212.694
11.300	570,300	8:3	2.634	2 634	2.419	2.100	1.334	.270				-221.283
	1030.000	2.720	2.7:4	2.5 6	2.314	2.915	1.245	. 260	.216	446		-228.500
	1056.036	2.754	E. 735	2.598	2.345	1.721	1.275	1.257	1.417	250		-240.605
	1110 00	2 577	5 0.2	:63	2.403	195	1.3.21	.762	1.749	1.152		240.918
	. 130.00	533	2.546	2 275	2.674	1.767	1.101	1.139	2.962	3.060		-281.284
	1320 030	2.117	3 128	2.871	2.4.6	1.678	932	. 196	3.269	5.417	4.749	-287.213
	4.3.036	2.53:	2.5%	2.415	3.000		1.137	2.232	6.597	11.877	14.541	-314.333
	96.66	- "90	2.€3:	2.803	2.403	1.770	1.025	3.654	11.570	21.321	25.845	-346.338
	1 300.000	2 534	2.513	2.3.0	1.904	2.176	249	4.903	13.899	24.990	28.695	-353.491
	1350.000	2.027		635	1.670	1.073	. 983	3.767	6.986	8.215	10.597	-33.972
	178.0 -0	1.650	1.032	1.698	0 AE	1.634	1.935	4.018	7.853	8.603	11.537	-36.472
35.446		107	1.1.3	1.151	1.144	1.142	.931	4.363	9.024	12.439	12.568	-49.099
1.5.010	250,000	1.129	1.152	1.169	1.4.75	1.177	1.031	5.577	9.848	13.783	15.164	-56.499
51, 110	-54.930	1.13	1.533	1.635	1.251	1.229	.933	6.113	11.044	15.169	17.293	-65.302
175 (8.	450 699	1.398	1.4.0	1 402	1.438	1.304	917	5.53é	10.903	16.001	18.242	-86.726
5.000	0.00	1 603	1.795	1.7.2	1.756	1 576	1.023	5.302	13.437	18.729	24.737	-118.981
15000	036.360	1.055	1.34E	1.610	1.670	1.571	.934	8.231	16.458	23.716	29 312	-122.763
	59.760	2.671	2.046	1.942	1.754	. 6 41	.900	9.336	13.290	27.184	35.827	-1.28.331
1.5.006	354.000	2.100	1.749	1.72	1.976	1.402	.677	11.844	23.834	37.369	48.305	-141.193
15 011	200 010	1.051	1.721	1.745	1.533	1.373	. 693	12.642	26.464	41.107	54.157	-140.161
135.000	950.000	2 142	1.585	1.750	1.531	1.337	.547	13.783	28.541	46.114		-149.068
125.000		2.145	1.972	1.733	1.518	1.344	.503	13.979	30.223	48.581	64.307	-144.658
	1.130.000	2.193	1.935	1.747	1.556	1.376	.521	15.237	32.542	52.120		-142.476
	1050.900	2.068	1.841	1.656	1.476	1.412	. 499	17.796	37.660	50.967	78.181	-146.828
135.000	1100.000	2.165	1.728	1.701	1.531	1.437	. 428	19.549	41.426	65.052		-143.833
	1150.000	1.963	1.746	1.538	1.475	1.465	.403	21.552	40.516	71.364		-144.614
35.000	1350.030	1.738	1.910	1.732	1.703	1.658	. 139	28.962	56.579	92.029		-138.916
135.000	1436.000	1.914	1.765	1.755	1.741	1.606	. 469	29.590	50.535	83.197		-143.369
135.000	1500.000	1.931	1.937	1.931	1.544	1.673	.571	35.681	55.179	90.482		-157.646
	1300.003	2.356	2.135	2.179	1.067	1.634	.764	37.473	67.397	96.759		-180.419
35.900	1850.000	: 996	2.026	2.100	1.902	1.505	.712	38.027	60.633	98.743	126.376	-186.972

CASE 3

THETA	FREQ	PI(1)	PI (2)	PI (3)	PI (4)	PI (5)	PI (6)	PH(2)	PH(3)	PH(4)	PH(5)	PH(6)
90.000	178.000	.998	1.013	1.013	1.922	1.007	1.967	5.456	10.953	16.134	20.195	696
99.000	260.000	. 259	.985	. 738	1.005	.938	1.041	6.329	12.514	17.755	23.167	1.546
70.000	250.006	.958	.994	1.0:4	1.016	1.053	.949	7.235	15.932	21.951	28.166	567
90.000	300.006	969	.995	1.008	1.005	1.018	. 982	8.623	17.300	25.518	32.264	-1.589
90.000	350.000	. 958	.935	1.013	1.024	1.039	.985	10.387	20.012	29.455	38.872	1.564
90.000	450.300	.973	1.01€	1.074	1.002	1.039	1.000	10.957	23.319	34.394	44.302	-2.030
*9.900	610.000	990	1.059	1.038	1.209	1.174	1.009	10.795	27.407	40.763	55.788	-6.534
90.000	550.300	1.025	1.067	1.119	1.138	1.145	1.607	15.400	31.564	47.252	62.538	. 964
96.000	750.000	1.030	1.150	1.131	1.130	1.179	1.2:8	17.723	35. 83	52.396	69.462	-1.588
.a 900	854.000	1.137	1.146	1.136	1.0 %	1.069	1.128	19.510	39.353	61.022	82.726	633
79.160	900.000	1.139	1.157	1.141	1.698	1.074	1.205	20.519	43.071	66.759	90.803	-3.103
26 636	953.606	1.103	1.138	1 1 2	135	070	1.176	22.193	44.047	69.906	94.896	466
20.000	70.600	1.170	1.203	1.157	1.193	1.080	1.230	21.743	44 376	71.748	98.477	- 120
	1090.30;	1.240	1.231	1 125	1.131	1 121	1.235	22.149	47.222	75.888	103.482	-1.498
	1050.000	1.154	1.132	1.0 21	1.045	1.135	1.226	26.152	54.986	85.169	113.704	304
	1100.000	1.236	1.179	1.131	1.160	1.139	1.247	25.900	55.315	38.138	119.029	922
	1153.030	1.121	:.051	036	1.034	1.1.22	1.178	28.534	63.125	99.007	129.207	. 322
	1356.300	1.394	1.055	1.039	1 32	1.246	1.139	33.004	76.514	:13.565	146.610	-5.567
	:400.066	. 905	. 949	1.044	1.17	1.259	1.043	42.592	84.755	121.476	152.113	-4.915
	1000.000	.996	1.134	1.236	1.254	1.261	1.020	48.596	37.903	125.909	165.183	-1.349
	:336.906	1 338	1.219	1.274	1.229	1.227	1.014	46.980	37.467	134.629	186.961	1.138
	1350.000	902	1.224	1.2.3	1.1.72	1.239	1.108	44.514	87.421	136.364	186.850	589
5 693	176.000	.930	.930	.972	.900	.950	1.986	3.750	8.564	14.554	17.075	32.589
15 000	200.900	.931	.977	.907	.973	.936	1.112	4.582	9,494	16.478	19.407	38.279
45.000	250.000	. 9.15	.934	. 935	.933	.957	1.110	5.983	13.966	13.714	29.078	47.957
4100	300.000	.905	.971	.952	.939	.937	1.142	6.765	13.741	21.411	30.498	54.747
45.000	250.000	. 764	. 262	.952	.928	.915	1.230	7.405	16.301	25.892	36.137	66.887
45.000	450.000	.937	.958	. 9.36	.374	. 8 36	1.433	9.572	22.553	33.823	48.269	94.740
45.000	ω:0.000	. 383	.8 *5	. 735	792	792	1.798	14.533	31.563	60.957	82.146	113.148
40.000	050.600	. 904	. 3 46	.734	710	.741	1.864	14.246	35.516	58.347	65.750	120.196
5.000	750.000	.787	. 044	. 337	.576	.639	2.239	20.944	50.696	34.997	112.908	131.221
15.006	657.000	.627	518	.514	.629	.752	2.131	32.719	72.207	105.698	128.974	136.386
45.046	900.006	.622	.514	.507	.029	.730	3.1.22	34.045	74.879	107.967	131.460	136.346
45.000	930.000	. 522	. 429	546	713	. 358	2.176	45.003	91.109	119.394	142.051	141.743
5.036	970.000	. 504	. 444	. 585	.750	. 387	2.205	43.336	39.305	117.308	139.102	-135.833 139.738
	1000.000	. 491	. 457	.622	.798	.905	2.237	51.010	67.171	116.824		
	1050.000	. 453	. 439	.010	.736	.869	2.174	55.523	94.439	119.281	145.910	145.209
	1106.000	.417	. 529	.740	.670	.971	2.209	56.503	68.199	111.962	133.535	138.510
	1150.000	. 373	.521	.729	.830	.033	2.339	54.425	95.172	108.402	131.679	
	1350.000	. 430	.705	. 9.25	.798	.722	2.030	32.028	55.930	63.439	123.450	136.606
	1409.006	. 508	.725	.361	.709	.629	1.961	31.814	56.505	87.001	172.131	156.066
	1500.000	. 530	.738	.637	.535	.672	1.921	25.821	56.124	112.106		
	1800.090	726	.734	. 533	.653	.915	2.037	28.744	79.983	151.023	193.278	186.887
45.000	1850.000	. 550	. 657	.535	.709	.870	2.963	32.900	97.172	147.573	173.162	107.012

CASE 3

THETA	FREQ	PI(1)	P1(2)	PI (3)	PI(4)	PI (5)	PI (6)	PH(2)	PH(3)	PH(4)	PH(5)	PH(6)
.000	178.000	.996	.987	.931	.961	.973	1.115	. 397	2.280	5.740	4.388	47.780
.000	200.000	1.009	1.014	1.012	1.008	. 984	1.160	1.040	2.423	6.240	3.791	54.284
.000	250.000	.925	.938	.940	.944	.927	1.228	1.462	3.476	4.183	10.606	67.854
.000	300.000	1.002	1.025	1.019	.975	.980	1.258	1.668	3.519	5.460	9.108	76.832
.000	350.000	1.027	1.049	1.047	1.018	.984	1.381	1.137	3.276	5.991	10.100	91.0BB
.000	450.000	1.076	1.140	1.105	1.073	1.001	1.712	1.177	4.620	7.642	13.177	113.250
.000	610.200	1.245	1.323	1.248	1.063	.944	2.267	3.704	6.039	15.753	23.095	150.830
. 000	650.000	1.264	1.308	1.224	1.115	.944	2.309	2.124	6.306	12.504	23.503	158.714
.000	750.000	1.322	1.301	1.200	1.017	.823	2.775	. 993	4.442	11.265	24.987	174.934
.009	950.000	1.264	1.235	1.111	. 906	.694	2.702	3.143	7.754	18.211	35.196	194.013
.000	700.000	1.206	1.133	1.057	. 948	.639	2.591	2.327	7.109	17.701	36.130	199.142
.000	950.000	1.313	1.283	1.119	.854	.570	2.934	2.312	7.289	18.421	42.250	212.672
.000	970.000	1.271	2.237	1.073	.810	.534	2.928	2.223	7.294	18.788	45.170	216.084
	1000.000	1.199	1.161	.997	.744	- 469	2.840	2.428	7.945	19.712	47.346	220.828
	1050.000	1.195	1.157	.979	.669	. 447	2.841	2.233	7.217	20.554	50.754	230.833
	1100.000	1.253	1.199	1.062	. 682	. 307	3.133	2.349	3.033	21.731	65.993	239.463
	1150.000	1.055	1.003	.003	.495	. 322	2.678	1.650	8.194	25.059	78.721	247.708
	1350.000	1.124	1.030	.769	. 335	. 344	3.280	3.956	12.164	40.726	123.534	279.886
	1 400.000	.915	.340	. 003	.272	.230	2.714	1.492	9.878	40.118	134.701	284.772
	1660.000	1.084	.941	. 577	. 228	.505	2.940	3.640	15.911	86.861	158.257	314.675
	1 300.030	.956	.812	. 461	.233	.547	2.531	6.880	22.891	121.262	171.509	345.703
	1350.000	1.220	1.012	.504	.248	.607	3.035	3.121	17.153	136.909	174.242	353.350
-45.000	176.000	.981	.991	.982	. 970	.988	1.050	-2.976	-5.027	-6.256	-11.161	35.498
-45.000	230.000	.997	1.015	1 025	1.036	1.025	1.076	-2.754	-5.387 -7.588	-6.259	-12.550 -13.771	39.229 50.288
-45.000	250.000	. 679	.997	1.014	.924	1.006	1.155	-4.206 -3.730	-7.503	-12.109	-14.969	57.868
45.030	350.016	.952	.994	1.019	1.013	1.014	1.210	-5.842	-10.408	-15.297	19.666	67.048
-45,000	456.000	.955	1.045	1.057	1.010	1.077	1.453	-7.957	-13.186	-19.945	-24.388	86.174
45.000	010.000	.913	1.101	1.155	1.1.22	1.131	1.828	-9.331	-17.394	-20.281	-27.998	115.746
-45.000	650.000	.873	1.969	1.142	1.199	1.167	1.331	-10.986	-17.660	-24.500	-29.997	122.186
-45.000	750.000	. 361	1.020	1.145	1.209	1.211	2.230	-13.549	-23.635	-31.355	-36.457	131.322
-45.000	350.001	.620	823	1.503	1.114	1.159	2.124	-16.430	-27.757	-35.253	-41.763	143.905
-45.000	900.000	. 537	.850	1.024	1.150	1.104	2.140	-19.016	-30.986	-40.112	-47.036	142.922
-45.000	250.000	.524	.739	1.035	1.173	1.220	2.260	-23.551	-36.951	-45.268	-49.487	149.762
-45.000	970.000	494	.745	.970	1.145	1.211	2.250	-28.243	-42.969	-52.231	-57.883	144.711
	1898. 30	.516	.753	. 299	1.104	1.220	2.293	-30.268	-46.456	-56.476	-63.334	144.558
	1050.000	. 471	.721	.935	1.156	1.302	2.103	-33.300	-49.817	-50.776	-67.301	150.490
	1100.000	434	.655	.957	1.178	1.220	2.279	-41.765	-62.402	-74.693	-01.757	145.069
	1150.000	. 386	.507	.871	1.005	1.161	2.381	-47.240	-66.903	-78.152	-84.196	140.246
	1350.000	.437	. 448	.812	1 . 1 49	1.257	2.125	-75.217	-110.378	-125.751	-133.066	140.320
	1400.000	. 453	. 431	.779	1.1.4	1.231	1.932	-79.179	-110.144		-135.148	147.187
	1600.000	.539	.279	. 522	1.066	1.235	1.957			-160.365	-169.036	159.536
	1300.000	.662	.373	. 440	1.051	1.313	2.102		-149.741			196.765
	1350.000	. 646	. 363	. 415	1.906	1.281	2.071		-153.657			198.439

CASE 3

THETA	FREQ	Pi(1)	P1(2)	Pi (3)	PI (4)	P1 (5)	PI (6)	PH(2)	PH(3)	PH(4)	PH(5)	PH(6)
-90.000	178.000	.987	.989	.987	.986	.976	.986	-4.482	-9.827	-14.143	-21.412	.7 =
-90.000	200.000	1.001	1.018	1.024	1.039	1.020	.938	-4.962	-9.814	-15.029	-23.4.79	391
	250.000	.900	.877	.872	.075	. 692	1.003	-6.815	-13.756	-22.545	-29.938	1.349
-90.000		. 940	.9-4	. 949	.954	.948	.903	-7.478	-15.041	-31.945	-33.121	1.07/
-90.000	300.000			. 762	.944	.930	.957	-9.150	-18.484	-29. 72	-48.717	40 !
-90.000	350.000	.953	. 958	. 951	.941	937	1.019	-14.203	-25.307	-46.536	-55.295	.110
-90.300	450.000	.951	.952	929	.927	.674	1.009	-19.914	-35.346	-49, 579	-73.573	3 580
-90.000	610.000	.987	959	.928	.691	874	1.019	17.710	-35.:04	-54 770	-78.627	- 147
-90.000	oS0.000	1.000		1.035	.954	.892	1.137	-17.041	-35.984	-50.157	04.409	2.165
-98.000	750.001	1.147	1.109	.927	. 839	.749	1.14:	-17.803	38.145	-63.77%	95 48:	1 . 3
-70.000	650.000	1.055	1.0.5	1.040	.947	.568	1.190	-19.035	- 17.9.7	-63.471	-91 754	1 500
-90.000	900.000	1.125	1.168		.052	.811	1.235	-19.819	-40.752	-66.765	182.436	3 . **
-90.000	950.000	1.149	5-114	1.951	.849	.779	1.264	-19.340	-39.629	64.695	-101.05:	01.7
-90.000	974.040	1.193	1.160		.945	.783	1.513	-19.339	-37.103	-61.550	-99.6:	2.100
	1060.000	1.240	1.233	1 - 1 3"		.760	1.234	-20.414	-41.330	-66.524	167. 4.3	950
	1950.000	1.172	1.190	1-116	1.002	.767	1.337	-21.315	-41.987	-67.239	100.65	21
	1100.000	1.205	1.266	1.174		. 796	1.176	-24.101	45.586	-69.043	184.157	1.781
	1150.000	1.119	1.1-33	1.150	.977	.627	1.182	-30.176	-54 (69		-112.031	1.764
	1350.000	1.128	1.272	1.315	1.143	. 6.24	032	-32 933	-56.475		-113. 69	1
	1400.000	1.016	1.213	1.295	1.175		1.967	-51.246			-139 561	.76 -
	1600.000	.750	1.119	1.3-4	1.373	1.942		-56.714		-132.696	61.370	.519
	1800.000	.967	.960	1.300	1.497	1.161	1.562	-56.637		-142.723	169.000	1.500
	1350.000	1.057	1.007	1.337	1.543	1.301	985	-3.161	-6.730	-1.3.832	-1842	-32,407
-135.000	178.559	1.049	1.050	1.045	1.037	.797	.975	-3.745	-7.952	-15.172	-29 715	-39.195
-135.000	200.000	1.670	1.030	874	1.0-7	1.039	.750	-4.255	-9.089	-15.568	-24.825	47.357
135.000	250.000	1.053	1.040	1.014	.967	.947	.756	-5.1.23	-11.457	-19.030	-27.837	-53.353
-135.000	300.000	1.696	1.036	1.065	1.314		. 976	5. 112	-1.2.003	-20.482	32.654	Sele . 750
135.000	350.000	1.194	1.165	1.139	1.975	1.981	1 921	-7.751	-1536	-26.711	-40.462	-05 74
135.000	450.000	1.364	1.337	1.2 ~	1.175	1.275	.780	-8 853	-16.427	-27 05	41 047	-116.717
-135.000	610.000	1.776	1.716	1.641	1.531		.924	-7.397	-13.771	-27 :19	42.097	-121.951
-135.000	654.000	1.626	1.796	1.711	1.521	1.306	.059	-0.311	-17.135	-28.1°A	-41.456	1 51 . 990
-135.000	750.000	2.102	2.145	2.074	1.895	1.016	443	-9,100	-18.922	-29.044	-45, 147	-1.36.042
135.000	054.810	2.009	2.074	2.001	1.6-31	1.626		-10.404	-28.97	-33.623	-47.821	139.375
-135.000	966.600	2.638	2.161	2.163	1.998	1.799	-650	-11.959	-21.964	-33.752	-51.6.5	
-135.000	930.000	2.126	2.292	2.300	2.121	1.830	.549	-11.484	-22.417	-34.153		-142.387
-135.000	970.000	2.145	2.327	2.367	2.188	1.330	.507	-12.299	-24.091	-35.270		-139.219
	1000.000	2.162	2.373	2.445	2.321	1.998	.487		-25.536	-39.510		-142.504
	1050.000	2.005	2.3.3	2.401	2.3.0	1.913	. 475	-13.295	-20.279	-42.675		-141.198
	1100.000	2.116	2.419	2.567	2.473	3.146	. 442					-141.583
-135.000	1150.000	1.998	2.276	2.431	2.356	2.072	. 441	-17.154	-31.523	-46.141		-133.57
-135.000	1350.000	2.017	2.349	2.642	2.600	2.447	. 444	-23.900	-42.674	-61.024	-81,939	-1.37.519
	1400.000	1.985	2.283	2.686	2.674	2.434	. 496	-25.306	-45.625		-109.289	
-135.000	1600.000	1.829	2.111	2.555	2.752	2.570	.585	-37.040	-62.315	-65.578		-183,005
-135.000	1800.000	1.766	2.203	2.703	3.002	2.931	.710	-39.607		-103.400		
-135.000	1850.000	1.937	2.139	2.678	3.021	2.070	. 685	-41.776	-77.114	-100.778	-133.937	-107.004

APPENDIX B

DERIVATION OF THE SCATTERING PARAMETER FOR THE CASE OF SPHERICAL ACOUSTIC WAVES

As long as the microphone itself is of finite size and its pressuresensing elements are separated by a finite distance, a pressure gradient
microphone having any arbitrary shape will be subject to three effects that
tend to impede its pressure gradient measurement capability. The most obvious
effect arises from the inherent need to approximate a gradient at a point
in space by a slope determined by finite differences. The other two effects
are related to scattering, and it is primarily for the purposes of clearer
analysis that they are considered as two distinct phenomena. One arises from
the fact that even at low frequency, pressure phase distortion takes place due
to the presence of the body. Finally, the last effect is attributable to
pressure amplitude and phase changes due to high frequency scattering.

The derivation of a scattering parameter σ which incorporates all three phenomena for the plane wave case has been previously presented in Refs. 1 and 2. This appendix is devoted to the derivation of an analogous expression for σ for the case where the distance between the sound source and scattering body is not large enough for the plane wave approximation to hold. In the experiments described in this report, the acoustic waves were considered to be spherical.

The complex pressure in a spherical wave field is assumed to be in the form

$$p_i = p_o \frac{e^{ikr}}{kr}$$

Note that the assumed sinusoidal time variation is suppressed. Let the center point of a cylindrical body having a face-to-face separation Δr be located a distance R from the acoustic source. It follows then that across the cylinder

$$\Delta p_i = \frac{p_0 e^{ikR}}{k} = \begin{pmatrix} ik \frac{\Delta r}{2} & -ik \frac{\Delta r}{2} \\ \frac{e}{R + \frac{\Delta r}{2}} & -\frac{e}{R - \frac{\Delta r}{2}} \end{pmatrix}$$

and

$$\frac{\Delta P_{i}}{\Delta r} = \left(\frac{\partial P_{i}}{\partial r}\right)_{r=R} = \left[\frac{R^{2} \left(\frac{ik \frac{\Delta r}{2}}{R + \frac{\Delta r}{2}} - \frac{e^{-ik \frac{\Delta r}{2}}}{R - \frac{\Delta r}{2}}\right)}{(ikR-1)\Delta r}\right]$$
(B-1)

In analogy with previous work in Refs. 1 and 2, the term in the square brackets can be viewed as the finite difference factor.

The body shape calibration factor K, accounting for low frequency phase distortion, is introduced through the definition

$$K = \lim_{ka \to 0} \left(\frac{\Delta P_1}{\Delta p} \right)$$
 (B-2)

where Ap is the measured pressure difference.

The effects of finite difference measurements and of low and high frequency scattering can be investigated by means of a scattering parameter σ , defined as

$$\sigma = \frac{\frac{\Delta p}{\Delta r} \frac{\Delta r (ikR-1)}{R^2 \left(\frac{e^{ik\Delta r/2}}{R+\Delta r/2} - \frac{e^{-ik\Delta r/2}}{R-\Delta r/2}\right)}{\left(\frac{\partial p_i}{\partial r}\right)_{r=R}}$$
(B-3)

After substitution for $\left(\frac{\partial p_i}{\partial r}\right)_{r=R}$, combining all complex terms, with the exception of Δp into the denominator, and evaluation of the absolute value, Eq. (B-3) assumes the form

$$\sigma = \frac{K k \left[R^2 - \left(\frac{\Delta r}{2}\right)^2\right] |\Delta p|}{2 p_0 \sqrt{R^2 \sin^2 \frac{\Delta r}{2} + \left(\frac{\Delta r}{2}\right)^2 \cos^2 k \frac{\Delta r}{2}}}$$

Finally, realizing that in terms of measurables

the expression for the scattering parameter can be written as

$$\sigma = \frac{K \left[R^2 - \left(\frac{\Delta \mathbf{r}}{2}\right)^2\right] \left|_{\Delta \mathbf{p}}\right|}{2 \left|_{\mathbf{p}_{\underline{\mathbf{i}}}}\right| R \sqrt{R^2 \sin^2 k \frac{\Delta \mathbf{r}}{2} + \left(\frac{\Delta \mathbf{r}}{2}\right)^2 \cos k \frac{\Delta \mathbf{r}}{2}}}$$
(B-4)

This is the expression which was used to compute scattering parameters from experimental data. It should be noted that since pressures are non-dimensionalized by the incoming pressure, no distinction need be made between maximum pressure amplitudes, or the rms values which are actually measured. The Δp difference, of course, takes into account the pressure phase differences.

In Fig. 21 σ is plotted on a decibel scale, i.e., the ordinate on the plot is 20 $\log \sigma$.

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15. Supplementary Notes			
Langley Technical Monitor: Final Report	: John M. Seiner		
Acoustic scattering on a of pressure gradient micromentally between ka value (L/D = 0.5 and 0.25) haviattached to preamplifiers cylindrical bodies, and f 38 cm diameter woofer in Surface pressure augmenta for various sound wave in retical predictions suppl results are tabulated in determined pressure augmetions. The agreement for exceptions. This is excean acoustic experiment of with ka and L/D ratio, as type of data represents a	ophones. These scattering of 0.407 and 4.232 using a 25 cm outside diametry by flexible connecturs, lush mounted on the extense large speaker enclosured tion and phase difference cidence angles. Results ied by NASA for ka = 0.40 the appendices. With minutations agreed to within relative phase angles willent, and approaches the type reported here. computed from experiment useful tool in the design	ng effects were ng two circular ter. Small con were installed for surface of was used as t es were compute are graphicall 07, 2.288, and nor exceptions, 10.75 dB with 15 within 5 per 16 realistic rep 17 Scattering pa 18 tal data, are a 18 nof pressure	investigated experi- cylindrical models denser microphones, from inside the the cylinders. A he sound source. d from measured data y compared with theo- 4.232. All other the experimentally theoretical predic- cent without any eatability limits in rameter variations lso presented. This
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